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CHOICE OF MEASURE SOURCE TERMS IN INTERFACE COUPLING FOR A MODEL PROBLEM IN GAS DYNAMICS

FRÉDÉRIC COQUEL*, EDWIGE GODLEWSKI†, KHALIL HADDAOUI‡, CLAUDE MARMIGNON§, AND FLORENT RENAC¶

Abstract. This paper is devoted to the mathematical and numerical analysis of a coupling procedure for one-dimensional Euler systems. The two systems have different closure laws and are coupled through a thin fixed interface. Following the work of [5], we propose to couple these systems by a bounded vector-valued Dirac measure, concentrated at the coupling interface, which in the applications may have a physical meaning. We show that the proposed framework allows to control the coupling conditions and we propose an approximate Riemann solver based on a relaxation approach preserving equilibrium solutions of the coupled problem. Numerical experiments in constrained optimization problems are then presented to assess the performances of the present method.

Key words. Model coupling, hyperbolic systems of conservation laws, measure source term, Riemann solver, relaxation, constrained optimization.

subject classifications. 35L04, 76M12, 76N15

1. Introduction

The study of large-scale and complex problems exhibiting a wide range of physical space and time scales (see for instance [62, 35, 14]), usually requires separate solvers adapted to the resolution of specific scales. This is the case of many industrial flows. Let us quote, for example, the numerical simulation of two-phase flows applied to the burning liquid oxygen-hydrogen gas in rocket engines [58]. This kind of flow contains both separated and dispersed two-phase flows, due to atomization and evaporation phenomena. This requires appropriate models and solvers for separated and dispersed phases that have to be appropriately coupled. Another example concerns turbomachine flows which can be modeled by the Euler equations of gas dynamics with different closure laws between the stages of the turbine, where the conditions of temperature and pressure are strongly heterogeneous. The coupling of these different systems is thus necessary to give a complete description of the flow inside the whole turbine.

The method of interface coupling allows to represent the evolution of such flows, where different models are separated by fixed interfaces. First, coupling conditions are specified at the interface to exchange information between the systems. The definition of transmission conditions generally results from physical consideration, *e.g.* the conservation or the continuity of given variables. Then, the transmission conditions are represented at the discrete level. The study of interface coupling for nonlinear hyperbolic systems has received attention for several years. In [43], the authors study the scalar case from both mathematical and numerical points of view.

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They couple two different one-dimensional conservation laws at a fixed interface. The case of system has been considered in [40], where admissible coupling conditions are defined at the interface. Indeed, since the systems we consider are hyperbolic, the boundary conditions must be imposed in a weak sense [29].

From previous studies, coupling conditions [43, 40, 10, 6, 50, 51, 7, 4, 39, 8, 52, 23, 18, 34, 56] can be classified in three categories: flux coupling, state coupling and coupling with measure source term. The flux coupling method is a conservative approach which ensures the continuity of the physical flux through the coupling interface. Conversely, the state coupling method is a non-conservative approach which imposes (at least weakly) the continuity property of either the (conservative) variables or a nonlinear transformation of them (say, primitive variables). Finally, the coupling condition can be modelled thanks to a bounded vector-valued Dirac measure concentrated at the coupling interface [40, 4, 5, 34]. The coupling condition is then prescribed from the definition of the mass of the measure. This modelling presents several advantages. It allows the consideration of both conservative and state couplings. Moreover, it enables the control of the coupling conditions by taking into account the mass, momentum or energy losses located through the discontinuity.

In this study, we focus on a model case which concerns the coupling of gas dynamics equations with different closure laws for the pressure. We propose an extension of the works [4, 34, 5] to the full Euler system (with energy) in the general case where all the source terms may be non zero. Indeed, in these previous contributions, the authors assumed the strict conservation of the density. The motivation of this present study is to extend the panel of applications, and the computations we present do involve test cases involving the energy equation which could not be covered in the previous work. These test cases are motivated by the applications among which the modeling of some sophisticated control devices meant to protect instruments from damages due to overheating. A few references on discharge relationships and component head losses for hydraulic structures or the modeling of pressure drop are already given in [5](see references [19, 34, 36, 37] therein). Some devices for the flow control involve the momentum and impact the energy conservation, note however that the topic of flow control is much broader and the simple models we are considering are but a small step in this direction for improving the simulation (see [32, 33] concerning aerospace for instance). One may also mention in the context of oil recovery the possibility of drawing out and reinjecting elsewhere a certain amount of gas (which is referred to as gas-lift and active bypass, see [30]), in order to prevent severe slugging, and this may be modeled as point losses in the mass conservation equation.

Indeed, an important remark is that the above mentioned devices are localized and small with respect to some characteristic length of the flow. Hence, in the modeling of the situation by a system of PDE, a natural scaling of the equations yields a point source at the interface, which means a multiple of a Dirac measure, classically placed at the origin. Following [37, 36], this can be interpreted as a scattering problem, with nonlinear waves interacting on a scale large compared to the scale of the interface which is then considered as infinitely thin.

Then, as explained in [37], the solution of the Riemann problem for the let us say simplified scaled model gives to the leading order the large time behavior of the full system with unscaled source. Thus, the study of Riemann problems comes naturally from the study of the interaction of nonlinear localized waves, and following this scattering setting, the Riemann solution represents the outgoing wave operator.

We show that the present framework is fairly flexible. In particular, it allows to

handle coupling problems constrained by the property that the Riemann solutions should keep their values in a given set while minimizing some convex nonlinear cost functions built on the source terms at the interface. These cost functions are defined by physical motivations.

Before we conclude this introduction, let us mention some very interesting works which, by some aspects, are strongly related to the present one. Interface coupling is clearly related to the study of conservation laws with discontinuous fluxes. In the scalar case, it is possible to give a rigorous framework in which existence and uniqueness results hold [11]. Then, singular source terms appear in different contexts which may correspond either to some (more or less) physical modelling as in [28] or [12, 13] or to numerical devices. Indeed, measure source terms are naturally introduced in many approaches for the construction of well balanced schemes for hyperbolic systems with geometric source terms [45] and, in turn, there is a natural link with the analysis of resonant systems [38] (and references therein). If general theoretical existence and uniqueness results may be obtained in the scalar case, for instance L^1 stability results in the non resonant case [49] or partial results in the resonant case [2], also for the existence of traces [61, 54], it is beyond the scope of the present work to give a rigorous notion of solution and well-posedness for the general coupling Cauchy problem in the case of systems. We rather focus on the solution of Riemann problems with coupling, which as mentioned above do represent at some scale wave the result of nonlinear wave interactions and moreover are naturally involved in the numerical approach, and on the optimization of source terms. Last, even if the link is not yet as close with interfacial coupling as the above subjects are, we should however mention the control of nonlinear hyperbolic systems [19] a subject appearing naturally with flow in networks [3].

The paper is organized as follows. The coupling problem is introduced in §2. The numerical approach to solve this coupling problem is described in §3. The method we consider relies on a relaxation approximation which is presented before the associated Riemann problem is solved. Then the source terms are chosen in order to preserve steady solutions of the coupled problem. Strategies for the dynamic evaluation of the source terms satisfying various criteria are then given. Several numerical experiments based on Riemann problems associated to the coupling model are then described in §4. Finally, the last section summarizes the conclusions of this study.

2. Coupling with measure source term

2.1. Physical problem and governing equations

We focus on the one-dimensional problem of interface coupling between two hyperbolic systems of conservation laws. More precisely, we consider the systems of Euler equations that model the flow of a compressible polytropic ideal gas. Both models are separated in space by a thin interface localized at $x=0$ and differ from their definition of the pressure. The flow is described by a system \mathcal{S}_L in the left domain $\mathcal{D}_L = \{x < 0, t > 0\}$ and by a system \mathcal{S}_R in the right domain $\mathcal{D}_R = \{x > 0, t > 0\}$. We look for a function $\mathbf{u} : (x, t) \in \mathbb{R} \times \mathbb{R}^+ \mapsto \mathbf{u}(x, t) \in \Omega \subset \mathbb{R}^3$, solution of:

$$\mathcal{S}_L : \quad \partial_t \mathbf{u} + \partial_x \mathbf{f}_L(\mathbf{u}) = 0, \quad (x, t) \in \mathcal{D}_L, \quad (2.1a)$$

$$\mathcal{S}_R : \quad \partial_t \mathbf{u} + \partial_x \mathbf{f}_R(\mathbf{u}) = 0, \quad (x, t) \in \mathcal{D}_R, \quad (2.1b)$$

with initial condition

$$\mathbf{u}(x, 0) = \mathbf{u}_0(x), \quad x \in \mathbb{R}.$$

The conservative variable vector and smooth flux functions are defined by:

$$\mathbf{u} = \begin{pmatrix} \rho \\ \rho u \\ \rho E \end{pmatrix} \quad \text{and} \quad \mathbf{f}_\alpha(\mathbf{u}) = \begin{pmatrix} \rho u \\ \rho u^2 + p_\alpha(\tau, \varepsilon) \\ \rho E u + p_\alpha(\tau, \varepsilon) u \end{pmatrix}, \quad \alpha = L, R.$$

The variables ρ , $\tau = 1/\rho$ and $\varepsilon = \varepsilon(\tau, s)$ stand for the density, the specific volume and the specific internal energy of the gas, respectively, where s denotes the physical entropy. The variable u denotes the velocity and $E = \varepsilon + u^2/2$ the specific total energy.

The thermodynamic variables are related by two ideal gas equations of state

$$p_\alpha(\tau, \varepsilon) = \frac{(\gamma_\alpha - 1)\varepsilon}{\tau}, \quad \alpha = L, R, \quad (2.2)$$

where γ_L and γ_R denote the respective ratio of specific heats (satisfying $\gamma_\alpha > 1$). Introducing the sound speed in the fluid

$$c_\alpha = c_\alpha(\tau, s) = \sqrt{\frac{\gamma_\alpha p_\alpha}{\rho}}, \quad \alpha = L, R, \quad (2.3)$$

the pressure laws are assumed to verify the following inequalities

$$p_\alpha > 0, \quad \left(\frac{\partial p_\alpha}{\partial \tau} \right)_s = -\rho^2 c_\alpha^2 < 0, \quad \left(\frac{\partial^2 p_\alpha}{\partial \tau^2} \right)_s > 0, \quad \alpha = L, R, \quad (2.4)$$

for all $\rho > 0$ and all s in \mathbb{R} .

The systems \mathcal{S}_L and \mathcal{S}_R are strictly hyperbolic over the following set of states

$$\Omega = \{ \mathbf{u} \in \mathbb{R}^3 : \rho > 0, u \in \mathbb{R}, \varepsilon > 0 \}.$$

More precisely, for $\alpha = L, R$, the system \mathcal{S}_α admits the three following eigenvalues

$$\lambda_1^\alpha(\mathbf{u}) = u - c_\alpha, \quad \lambda_2^\alpha(\mathbf{u}) = u, \quad \lambda_3^\alpha(\mathbf{u}) = u + c_\alpha,$$

which are real and distinct. The characteristic field associated with λ_2^α is linearly degenerate, while the characteristic fields associated with λ_1^α and λ_3^α are genuinely nonlinear.

2.2. Coupling model

Following [4, 5], and [34], we propose here to model the coupling between systems (2.1) thanks to a bounded vector-valued Dirac measure concentrated at the coupling interface $x = 0$. Let us consider an application of the form $t \in \mathbb{R}_+^* \mapsto \mathcal{M}(t) = (\mathcal{M}_\rho(t), \mathcal{M}_{\rho u}(t), \mathcal{M}_{\rho E}(t))^T \in \mathbb{R}^3$; we then look for a solution to the following initial value problem

$$\partial_t \mathbf{u} + \partial_x \mathbf{f}(\mathbf{u}, x) = \mathcal{M}(t) \delta_{x=0}, \quad x \in \mathbb{R}, t > 0, \quad (2.5a)$$

$$\mathbf{u}(x, 0) = \mathbf{u}_0(x), \quad x \in \mathbb{R}, \quad (2.5b)$$

where $\delta_{x=0}$ denotes the Dirac delta function. The components of \mathcal{M} represent respectively the losses of mass, momentum and energy through the coupling interface. The discontinuous flux function introduced in (2.5a) is defined by:

$$\mathbf{f}(\mathbf{u}, x) = \begin{cases} \mathbf{f}_L(\mathbf{u}), & x < 0, \\ \mathbf{f}_R(\mathbf{u}), & x > 0, \end{cases}$$

where the flux depends on x through the pressure law

$$p(\tau, \varepsilon, x) = \begin{cases} p_L(\tau, \varepsilon), & x < 0, \\ p_R(\tau, \varepsilon), & x > 0. \end{cases} \quad (2.6)$$

Using standard arguments (see for instance [41]), any solution to the coupling problem (2.5) is subject to the following coupling conditions through the interface:

$$\mathbf{f}_R(\mathbf{u}(0^+, t)) - \mathbf{f}_L(\mathbf{u}(0^-, t)) = \mathcal{M}(t), \quad t > 0, \quad (2.7)$$

where $\mathbf{u}(0^-, t)$ and $\mathbf{u}(0^+, t)$ denote the left and right traces of the solution \mathbf{u} at time t . This model allows the application of various coupling strategies. Indeed, the weight $\mathcal{M}(t)$ can be seen as a user parameter, whose definition results generally from physical motivations. For instance, the definition of a weight $\mathcal{M} \equiv 0$ allows to ensure the conservation of \mathbf{u} over the whole space domain. Conversely, a state coupling method with transmission of a set of variables may lead to the definition of a non zero weight. Given an admissible change of variables $\mathbf{v} = \varphi_\alpha(\mathbf{u})$, $\alpha = L, R$, the state \mathbf{v} can be transmitted through the coupling interface by setting $\varphi_L(\mathbf{u}(0^-, t)) = \varphi_R(\mathbf{u}(0^+, t))$, $t > 0$ (see [8] for more details). For example, the transmission of primitive variables ρ , u and p through this interface results from $\varphi_\alpha(\rho, \rho u, \rho E) := (\rho, u, p_\alpha(\tau, \varepsilon))$, $\alpha = L, R$. The present contribution investigates the general case where all the weights may be non zero.

Note that (2.5a) can be written equivalently

$$\begin{cases} \partial_t \mathbf{u} + \partial_x \mathbf{F}(\mathbf{u}, H) - \mathcal{M}(t) \partial_x H = \mathbf{0}, & x \in \mathbb{R}, t > 0, \\ \partial_t H = 0, \end{cases} \quad (2.8)$$

where $H = H_{x=0}$ is the Heaviside function and $\mathbf{F}(\mathbf{u}, H) = (1 - H)\mathbf{f}_L(\mathbf{u}) + H\mathbf{f}_R(\mathbf{u})$. Thus we get a non conservative system in variables (\mathbf{u}, H) , for which the initial data is (2.5b) for \mathbf{u} augmented by $H(x, 0) = H(x)$. The matrix of (2.8) writes

$$M(\mathbf{u}, H) = \begin{pmatrix} \partial_{\mathbf{u}} \mathbf{F} & \mathbf{f}_R - \mathbf{f}_L - \mathcal{M}(t) \\ 0 & 0 \end{pmatrix}$$

with $\partial_{\mathbf{u}} \mathbf{F}(\mathbf{u}, H) = (1 - H)\mathbf{f}'_L(\mathbf{u}) + H\mathbf{f}'_R(\mathbf{u})$. This shows clearly that the presence of the measure source terms adds to the initial waves, associated to the eigenvalues of $\mathbf{f}'_{L/R}(\mathbf{u})$, a stationary discontinuity at $x = 0$. The system is resonant if a genuinely nonlinear field (thus associated to an acoustic wave) superimposes this discontinuity, we will thus restrict ourselves to the study of subsonic flows; this assumption prevents this resonance phenomenon which brings non uniqueness.

3. Numerical method

The numerical method we present will eventually result in a classical algorithm which can be implemented as section 4 proves. It is however interesting to understand the main elements which guided us to its derivation, since the scheme properties are obviously inherited from this construction.

3.1. Relaxation approximation

We are interested in solving numerically (2.5). In the context of finite volume methods, the Godunov method requires the exact solution of Riemann problems associated to (2.5) and may be expensive because of the presence of nonlinearities due

to the pressure law. The relaxation approach we will follow is an attempt to reduce the computational cost, still involving the Godunov method but on a simpler system approximating (2.5). It consists in the construction of a linear or quasilinear enlarged system, with only linearly degenerate fields, involving a source term which ensures that, in the regime of large values of a relaxation parameter $\lambda > 0$, the solution of this system converges to the solution of the original system. We refer the reader to [55, 24, 17] for discussions on the relaxation approach in the approximation of hyperbolic systems of conservation laws, [53, 16] for relaxation numerical methods in a general setting, or [20, 22, 25, 26, 27, 60, 59, 9] for precise examples in the context of fluid flows.

Let us introduce the following system

$$\partial_t \mathbf{U}_\lambda + \partial_x \mathbf{F}(\mathbf{U}_\lambda) = \lambda \mathcal{R}(\mathbf{U}_\lambda, x) + \widetilde{\mathcal{M}}(t) \delta_{x=0}, \quad x \in \mathbb{R}, t > 0, \quad (3.1)$$

with

$$\mathbf{F}(\mathbf{U}) = \begin{pmatrix} \rho u \\ \rho u^2 + \pi \\ \rho E u + \pi u \\ \rho \pi u + a^2 u \end{pmatrix}, \quad \mathcal{R}(\mathbf{U}, x) = \begin{pmatrix} 0 \\ 0 \\ 0 \\ \rho(p(\tau, \varepsilon, x) - \pi) \end{pmatrix}, \quad \widetilde{\mathcal{M}}(t) = \begin{pmatrix} \mathcal{M}(t) \\ \mathcal{M}_{\rho\pi}(t) \end{pmatrix},$$

where $a > 0$ is a given constant which will be specified below. We have introduced a new variable π which can be seen as a linearization of the original pressure p and is called the relaxed pressure; in the source term \mathcal{R} , the pressure p is defined by (2.6). With some abuse of notation, we note $\mathbf{U} = (\mathbf{u}, \rho\pi)$ (the rigorous notation should be $\mathbf{U} = (\mathbf{u}^T, \rho\pi)^T$), the vector of variables \mathbf{U} is assumed to be defined on the set of states $\Omega^r = \Omega \times \mathbb{R}$ and we introduce Π the projection operator $\Pi: \mathbb{R}^4 \rightarrow \mathbb{R}^3$ such that $\Pi \mathbf{U} = \mathbf{u}$. Formally, we can observe that the projection $\Pi \mathbf{U}_\lambda$ of the solution of the relaxation system (3.1) converges to a solution of the original system (2.5) (this last system is thus called the *equilibrium* system) when the relaxation parameter λ goes to infinity since the fourth equation in (3.1) imposes

$$\lim_{\lambda \rightarrow \infty} \pi_\lambda = p,$$

(at least formally).

The time dependent weight $\mathcal{M}_{\rho\pi}(t) \in \mathbb{R}$ corresponds to a new degree of freedom and will be defined later on.

For ease of notation, we now omit the subscript λ . The homogeneous system (3.1) with $\lambda = 0$ and $\widetilde{\mathcal{M}} \equiv 0$ is hyperbolic on Ω^r and possesses the eigenvalues

$$\lambda_1^r(\mathbf{U}) = u - a\tau, \quad \lambda_2^r(\mathbf{U}) = u, \quad \lambda_3^r(\mathbf{U}) = u + a\tau,$$

where the eigenvalue λ_2^r is double. All the characteristic fields associated with these eigenvalues are linearly degenerate. As seen when writing system (2.8), the presence of the source term in (3.1) results in an additional stationary discontinuity, which, with some abuse, we also call a contact wave with speed $\lambda_0^r(\mathbf{U}) = 0$ located at the coupling interface. The coefficient a has the dimension of a Lagrangian sound speed, and appears as a linearization of ρc . In order to prevent system (3.1) from instabilities in the regime of large values of λ , this coefficient a must verify the so-called subcharacteristic condition

$$a > \max(\rho c_\alpha), \quad \alpha = L, R, \quad (3.2)$$

for all the states under consideration. This condition expresses the interlacing of the eigenvalues of the relaxation and equilibrium systems $u - \tau a < u - c_\alpha < u < u + c_\alpha < u + \tau a$. We refer the reader to [17] for a rigorous study of stability conditions.

Any solution to (3.1) with initial condition (and a finite relaxation rate $\lambda > 0$), is subject to the following coupling condition through the coupling interface

$$\mathbf{F}(\mathbf{U}(0^+, t)) - \mathbf{F}(\mathbf{U}(0^-, t)) = \widetilde{\mathcal{M}}(t), \quad t > 0. \quad (3.3)$$

Finally, note that the flux $\mathbf{F}(\mathbf{U})$ in (3.1) is now globally defined and does not depend on the space variable, contrary to the flux $\mathbf{f}(\mathbf{u}, x)$ in (2.5).

3.2. The main features of the scheme

The approach we follow involves an enlarged system (3.1) with two different source terms. The first one corresponds to relaxation and the second one to the original point source term. The items are linked in that the relaxation procedure is introduced as a practical mean to ensure the well balanced property, i.e., to preserve the stationary solutions associated to the point source. However, the two source terms are not treated in the same way in the time discretization which involves a time splitting technique.

Let Δt be a time step, and set $t^{(n)} = n\Delta t, n \geq 0$. Using a time splitting technique, the first step concerns an enlarged system, one solves (3.1) with $\lambda = 0$ on an interval of length Δt , say $]t^{(n)}, t^{(n+1)}]$, which allows solutions away from equilibrium but ensures the well balanced property. Relaxation towards equilibrium is performed instantaneously in a second step, which amounts to solve

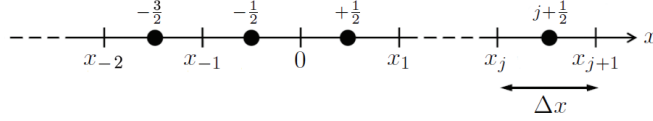
$$\partial_t \mathbf{U}_\lambda = \lambda \mathcal{R}(\mathbf{U}_\lambda, x) \quad (3.4)$$

on $]t^{(n)}, t^{(n+1)}]$, starting with the data, noted $\mathbf{U}_\lambda(t^{(n+1)-})$, given by the solution of the first step at time $t^{(n+1)}$, and let $\lambda \rightarrow \infty$. In the time discretization of this step, we use a simple implicit Euler method, so that $\mathbf{u}_\lambda(t^{(n+1)}) = \mathbf{u}_\lambda(t^{(n+1)-})$ and as $\lambda \rightarrow \infty$, $\rho(p - \pi)(t^{(n+1)}, x) \rightarrow 0$.

For our fully discrete scheme, in which well balanced and time stepping techniques complete each other, several ingredients are needed, they are described precisely in the following subsections. Let us make first two comments ((i) and (ii) below) because the similarity between the two source terms, similarity pointed out in [45], may however introduce some confusion. Indeed, the treatment of the relaxation source term (in the limit $\lambda \rightarrow \infty$) is responsible for discontinuity in the approximate solution at each time $t^{(n+1)-}$. While the well balanced property is achieved on the ground of a geometric source term taking the form of a Dirac measure in space. We detail these points in paragraph (ii). Moreover, both relaxation and point source terms involve some notions of *equilibria*. We must however distinguish the underlying notions which we do in the following lines.

- (i) Concerning relaxation, the first notion of equilibrium refers to instantaneous relaxation. The formal limit, letting the relaxation time λ^{-1} goes to 0 in (3.1), yields the *equilibrium system* (2.5). Associated to this latter system are the equilibria states $\mathbf{U}^{eq} = (\mathbf{u}, \rho p)$. This naturally defines an operator $\mathbf{U} \rightarrow \mathbf{U}^{eq}$ which we note \mathcal{E} . This operator is involved in the second step of the splitting method and precised in section 3.4.

The second notion of equilibrium refers to the well balanced property, which is often defined as *preserving equilibria*. In this context, equilibria are thus stationary solutions of the PDE with point source (cf. Definition 3.1 below). They are analyzed in section 3.5.

FIG. 3.1. *Space mesh.*

- (ii) The treatment of the relaxation source term by a time splitting procedure is responsible for a discontinuity wrt. the time variable, for all $x \in \mathbb{R}$. Indeed, following L. Gosse in [46] (who refers to a prior work by F. Bouchut [15]), the classical time-splitting discretization of (3.1) leads to write

$$\partial_t \mathbf{U} + \partial_x \mathbf{F}(\mathbf{U}) = \sum_{n>0} (\mathcal{E}(\mathbf{U}^{(n)-}) - \mathbf{U}^{(n)-}) \delta_{t=t_n} + \widetilde{\mathcal{M}}(t) \delta_{x=0}, \quad x \in \mathbb{R}, t > 0, \quad (3.5)$$

where $\delta_{t=t^{(n)}}$ denotes the Dirac measure at point $t^{(n)}$, acting on functions of the time variable. We have noted $\mathbf{U}^{(n)-} = \mathbf{U}(\cdot, t^{(n)-})$ the limit of $\mathbf{U}(\cdot, t)$ as $t \rightarrow t^{(n)}$, with $t < t^{(n)}$. The added delta measure terms may be seen as coming from the discretization of the time variable, approximating t by a step function equal to $t^{(n)}$ on $[t^{(n)}, t^{(n+1)})$. This amounts to approximate $1 = \partial_t t$ by a sum $\Delta t \sum_n \delta_{t=t^{(n)}}$ of Dirac measures at points $t^{(n)}$ with weight Δt . This approximation mimicks the pioneering approach of [48] for the derivation of well balanced schemes. Note that the approach in [48] considered an equation with geometric source term, which thus concerned the space variable. Then, the discretization of the singular relaxation source term $\lambda \mathcal{R}(\mathbf{U})$ in the limit $\lambda \rightarrow \infty$ takes the form $\sum_{n>0} (\mathcal{E}(\mathbf{U}^{(n)-}) - \mathbf{U}^{(n)-}) \delta_{t=t_n}$. Formally, in the numerical treatment of (3.4), the relaxation time $\frac{1}{\lambda}$ is replaced by a time step Δt .

We now give the great lines of our fully discretized numerical method. In the above described time splitting algorithm, only the first step needs a space discretization. We use a Godunov type scheme with an exact Riemann solver for the enlarged system ((3.1) with $\lambda=0$) which is introduced in section 3.3. Thus we need to solve the Riemann problem, which is performed in section 3.6 (uniqueness requires we are away from resonance, a condition which is fulfilled by assuming that the flow is subsonic). We will see that adding the energy equation, while assuming non zero weights, renders the computations trickier than in [5]. The relaxation procedure adds a weight $\mathcal{M}_{\rho\pi}$ which may be chosen (see section 3.7) in order to preserve stationary solutions (defined in section 3.5).

3.3. Godunov-type scheme

We aim at solving numerically the coupling problem (2.5) by a finite volume method, and first introduce some notations. Let Δx be the space step which we take uniform for simplicity. The nodes of the spatial mesh are defined by $x_{j+\frac{1}{2}} = \frac{1}{2}(x_j + x_{j+1})$, $j \in \mathbb{Z}$ where the points $x_j = j\Delta x$ correspond to the interfaces between elements $\mathcal{C}_{j+\frac{1}{2}} = (x_j, x_{j+1})$. The coupling interface is located in $x_0 = 0$ and the points x_j are in the left (resp. right) domain if $j < 0$ (resp. $j > 0$) (see figure 3.1).

At each time step $t^{(n)}$, we look for a piecewise constant numerical solution $\mathbf{u}_\Delta(x, t^{(n)})$ of the form:

$$\mathbf{u}_\Delta(x, t^{(n)}) = \mathbf{u}_{j+\frac{1}{2}}^{(n)}, \quad \forall x \in \mathcal{C}_{j+\frac{1}{2}}.$$

For $t=0$, the initial condition reads

$$\mathbf{u}_{j+\frac{1}{2}}^{(0)} = \frac{1}{\Delta x} \int_{c_{j+\frac{1}{2}}} \mathbf{u}_0(x) dx, \quad \forall j \in \mathbb{Z}.$$

For simplicity, we consider a three point finite volume scheme of the form:

$$\mathbf{u}_{j-\frac{1}{2}}^{(n+1)} = \mathbf{u}_{j-\frac{1}{2}}^{(n)} - \frac{\Delta t}{\Delta x} ((\mathbf{g}_L)_j^n - (\mathbf{g}_L)_{j-1}^n), \quad j \leq 0, \quad (3.6)$$

$$\mathbf{u}_{j+\frac{1}{2}}^{(n+1)} = \mathbf{u}_{j+\frac{1}{2}}^{(n)} - \frac{\Delta t}{\Delta x} ((\mathbf{g}_R)_{j+1}^n - (\mathbf{g}_R)_j^n), \quad j \geq 0, \quad (3.7)$$

where the numerical flux

$$(\mathbf{g}_\alpha)_j^n = \mathbf{g}_\alpha \left(\mathbf{u}_{j-\frac{1}{2}}^{(n)}, \mathbf{u}_{j+\frac{1}{2}}^{(n)} \right), \quad \alpha = L, R,$$

corresponds to a consistent approximation of the exact flux \mathbf{f}_α and is obtained by an approximate Riemann solver. In order to construct this solver, we use the relaxation approach and consider system (3.1). At each time iteration, Riemann problems must be considered:

- a Riemann problem defined at any internal interface $x_j, j \neq 0$:

$$\begin{cases} \partial_t \mathbf{U} + \partial_x \mathbf{F}(\mathbf{U}) = 0, & \forall x \in (x_{j-\frac{1}{2}}, x_{j+\frac{1}{2}}), \quad t \in (t^{(n)}, t^{(n+1)}], \\ \mathbf{U}(x, t^{(n)}) = \begin{cases} \mathbf{U}_{j-\frac{1}{2}}^{(n)}, & x < x_j, \\ \mathbf{U}_{j+\frac{1}{2}}^{(n)}, & x > x_j, \end{cases} \end{cases}$$

- a Riemann problem with source term at the coupling interface $x_0=0$:

$$\begin{cases} \partial_t \mathbf{U} + \partial_x \mathbf{F}(\mathbf{U}) = \widetilde{\mathcal{M}}^{(n)} \delta_{x=0}, & \forall x \in (x_{-\frac{1}{2}}, x_{+\frac{1}{2}}), \quad t \in (t^{(n)}, t^{(n+1)}], \\ \mathbf{U}(x, t^{(n)}) = \begin{cases} \mathbf{U}_{-\frac{1}{2}}^{(n)}, & x < 0, \\ \mathbf{U}_{+\frac{1}{2}}^{(n)}, & x > 0, \end{cases} \end{cases}$$

where $\widetilde{\mathcal{M}}^{(n)}$ denotes some consistent discretization of $\frac{1}{\Delta t} \int_{t^{(n)}}^{t^{(n+1)}} \widetilde{\mathcal{M}}(s) ds$, the average weight on $(t^{(n)}, t^{(n+1)})$. In the above Riemann problems, for both cases we have defined the data $\mathbf{U}_{j\pm\frac{1}{2}}^{(n)}$ as:

$$\mathbf{U}_{j-\frac{1}{2}}^{(n)} = \left(\mathbf{u}_{j-\frac{1}{2}}^{(n)}, \rho_{j-\frac{1}{2}}^{(n)} p_L \left(\tau_{j-\frac{1}{2}}^{(n)}, \varepsilon_{j-\frac{1}{2}}^{(n)} \right) \right) \quad \text{if } j \leq 0, \quad (3.8)$$

$$\mathbf{U}_{j+\frac{1}{2}}^{(n)} = \left(\mathbf{u}_{j+\frac{1}{2}}^{(n)}, \rho_{j+\frac{1}{2}}^{(n)} p_R \left(\tau_{j+\frac{1}{2}}^{(n)}, \varepsilon_{j+\frac{1}{2}}^{(n)} \right) \right) \quad \text{if } j \geq 0, \quad (3.9)$$

which means we have completed the state $\mathbf{u}_{j\pm\frac{1}{2}}^{(n)}$ by a relaxed pressure at equilibrium, so that all the data $\mathbf{U}_{k+\frac{1}{2}}^{(n)}, k \in \mathbb{Z}$ are at equilibrium.

3.4. Details of the time splitting method

In order to define the piecewise constant $\mathbf{u}_\Delta(x, t^{(n)}) = (\rho_\Delta, \rho u_\Delta, \rho E_\Delta)^T$ approximating the solution of (2.5) at time $t^{(n+1)}$, we proceed in two steps.

- *First step: evolution in time* ($t^{(n)} \rightarrow t^{(n+1)-}$). We solve numerically the system (3.1) with $\lambda=0$. In this first step, we consider the Cauchy problem

$$\begin{cases} \partial_t \mathbf{U} + \partial_x \mathbf{F}(\mathbf{U}) = \widetilde{\mathcal{M}}^{(n)} \delta_{x=0}, & x \in \mathbb{R}, \quad t \in (t^{(n)}, t^{(n+1)}], \\ \mathbf{U}(x, t^{(n)}) = \mathbf{U}_\Delta(x, t^{(n)}), & x \in \mathbb{R}, \end{cases} \quad (3.10)$$

with

$$\mathbf{U}_\Delta(x, t^{(n)}) = (\mathbf{u}_\Delta(x, t^{(n)}), (\rho\pi)_\Delta(x, t^{(n)})), \quad (3.11)$$

and where $\rho\pi$ verifies the equilibrium condition

$$(\rho\pi)_\Delta(x, t^{(n)}) = \rho_\Delta(x, t^{(n)}) p(\tau_\Delta(x, t^{(n)}), \varepsilon_\Delta(x, t^{(n)}), x).$$

The exact solution to (3.10) is obtained by juxtaposition of the solutions of the local Riemann problems set at each interface separating two adjacent elements. These solutions do not interact during the period Δt if the space and time steps are linked by the usual Courant-Friedrich-Lewy condition:

$$\frac{\Delta t}{\Delta x} \max_{\mathbf{U}} |\lambda_i^r(\mathbf{U})| < \frac{1}{2}, \quad i=0,1,2,3, \quad (3.12)$$

where the maximum is taken over all the states \mathbf{U} under consideration. We note $\mathbf{U}(x, t^{(n+1)-})$ the solution obtained at the end of this step.

- *Second step: instantaneous relaxation* ($t^{(n+1)-} \rightarrow t^{(n+1)}$). In this step, we solve the following initial value problem:

$$\begin{cases} \partial_t \mathbf{U} = \lambda \mathcal{R}(\mathbf{U}, x), & x \in \mathbb{R}, \quad t \in (t^{(n)}, t^{(n+1)}], \\ \mathbf{U}(x, t^{(n)}) = \mathbf{U}_\Delta(x, t^{(n+1)-}), & x \in \mathbb{R}, \end{cases} \quad (3.13)$$

in the asymptotic regime $\lambda \rightarrow \infty$. This amounts to impose that the function

$$\mathbf{U}_\Delta(x, t^{(n+1)}) = (\mathbf{u}_\Delta(x, t^{(n+1)}), (\rho\pi)_\Delta(x, t^{(n+1)})), \quad (3.14)$$

has the following components

$$\begin{aligned} \mathbf{u}_\Delta(x, t^{(n+1)}) &= \mathbf{u}_\Delta(x, t^{(n+1)-}), \\ (\rho\pi)_\Delta(x, t^{(n+1)}) &= \rho_\Delta(x, t^{(n+1)-}) p(\tau_\Delta(x, t^{(n+1)-}), \varepsilon_\Delta(x, t^{(n+1)-}), x). \end{aligned}$$

As usual these two steps are followed by a projection of $\mathbf{u}_\Delta(x, t^{(n+1)})$ on piecewise constant functions in order to define $\mathbf{u}_{j+\frac{1}{2}}^{(n+1)}$.

3.5. Equilibrium solutions

Following [34], given a weight $\mathcal{M} = (\mathcal{M}_\rho, \mathcal{M}_{\rho u}, \mathcal{M}_{\rho E})^T$, we look for a definition of the weight $\mathcal{M}_{\rho\pi}$ such that the above relaxation approach preserves the stationary solutions of the original coupling problem (2.5) in the sense of the following definition.

DEFINITION 3.1. *Let $\mathcal{M} = (\mathcal{M}_\rho, \mathcal{M}_{\rho u}, \mathcal{M}_{\rho E})^T$ be a constant weight and $\mathbf{u}_L, \mathbf{u}_R$ two constant states belonging to the space of states Ω such that*

$$\mathbf{f}_R(\mathbf{u}_R) - \mathbf{f}_L(\mathbf{u}_L) = \mathcal{M}. \quad (3.15)$$

Then, the function defined by

$$\mathbf{u}(x, t) = \begin{cases} \mathbf{u}_L, & x < 0, t > 0, \\ \mathbf{u}_R, & x > 0, t > 0, \end{cases}$$

is called an equilibrium solution for the original coupling problem (2.5).

Let us note that the stationary solutions of the coupled Riemann problem are an important tool in the study of solutions to conservation laws with discontinuous flux since they play the role of constants in the classical situation of a smooth flux (see [11] and references therein) and can thus be used in Kruzhkov type entropies; they are called *elementary solutions* and the set of such states gives birth to the notion of *germs*. The theory for systems with discontinuous flux is not as complete. However, the above definition is a natural extension to systems with source term (hence to the non-conservative problem (2.5)). We will require our scheme to be *well balanced* in the sense that it should preserve these equilibria (we refer to [21] for some general considerations on well balanced simple scheme). For a conservative system with a continuous flux, constants are elementary solutions, a conservative scheme with consistent flux preserves constant states. One simply imposes that there is only one numerical flux at each interface, and that this flux coincides with the exact flux if the states on each side are equal.

3.6. Riemann problem associated to the relaxation system

The first step of the splitting procedure in §3.4 requires the solution to the Riemann problem associated to the system (3.1) with $\lambda=0$. In this paper, we restrict ourselves to the study of subsonic flows. Given two subsonic constant states \mathbf{U}_L and \mathbf{U}_R in Ω^r , we solve the following initial value problem

$$\begin{cases} \partial_t \mathbf{U} + \partial_x \mathbf{F}(\mathbf{U}) = \widetilde{\mathcal{M}} \delta_{x=0}, & x \in \mathbb{R}, t > 0, \\ \mathbf{U}(x, 0) = \begin{cases} \mathbf{U}_L, & x < 0, \\ \mathbf{U}_R, & x > 0, \end{cases} \end{cases} \quad (3.16)$$

together with the coupling condition (3.3), where $\widetilde{\mathcal{M}}$ does not depend on time to ensure self similarity in the solution.

The solution to (3.16) contains only contact discontinuities since all the fields associated with the eigenvalues of the system (3.1) (with $\lambda=0$) are linearly degenerate. Moreover, the Whitham condition (3.2) associated to the subsonic flow hypothesis lead us to consider two wave structures for the solution to (3.16). As shown in figure 3.2, the intermediate states are denoted $\mathbf{U}_L, \mathbf{U}_1, \mathbf{U}_-, \mathbf{U}_+, \mathbf{U}_R$ when $\lambda_2^r < 0$ and $\mathbf{U}_L, \mathbf{U}_-, \mathbf{U}_+, \mathbf{U}_3, \mathbf{U}_R$ when $\lambda_2^r > 0$. In particular, we note \mathbf{U}_\pm the traces $\mathcal{W}(0\pm; \mathbf{U}_L, \mathbf{U}_R, \widetilde{\mathcal{M}})$ at the interface $x = \lambda_0^r = 0$. In order to respect these structures, the weights $\mathcal{M}_\rho, \mathcal{M}_{\rho u}, \mathcal{M}_{\rho E}$ and $\mathcal{M}_{\rho \pi}$ must be defined in an admissibility domain $\mathcal{D}_{adm}(\mathbf{U}_L, \mathbf{U}_R) = \mathcal{D}_{adm}^-(\mathbf{U}_L, \mathbf{U}_R) \cup \mathcal{D}_{adm}^+(\mathbf{U}_L, \mathbf{U}_R)$. The definition of this domain is linked to the fact that, on the one hand, these weights must respect the subsonic structure of the solution and, on the other hand, the states are constrained to remain in the space of states Ω^r . More precisely, using obvious notations for the components of the intermediate states, we define implicitly the admissible sets by

$$\mathcal{D}_{adm}^- = \left\{ \widetilde{\mathcal{M}} \in \mathbb{R}^4 : (\lambda_1^r, -\tau_1, u_-, -\tau_-, -\tau_+, -\lambda_3^r, -\varepsilon_1, -\varepsilon_-, -\varepsilon_+) (\widetilde{\mathcal{M}}) < 0 \right\}, \quad (3.17)$$

$$\mathcal{D}_{adm}^+ = \left\{ \widetilde{\mathcal{M}} \in \mathbb{R}^4 : (\lambda_1^r, -\tau_3, -u_+, -\tau_-, -\tau_+, -\lambda_3^r, -\varepsilon_3, -\varepsilon_-, -\varepsilon_+) (\widetilde{\mathcal{M}}) < 0 \right\}. \quad (3.18)$$

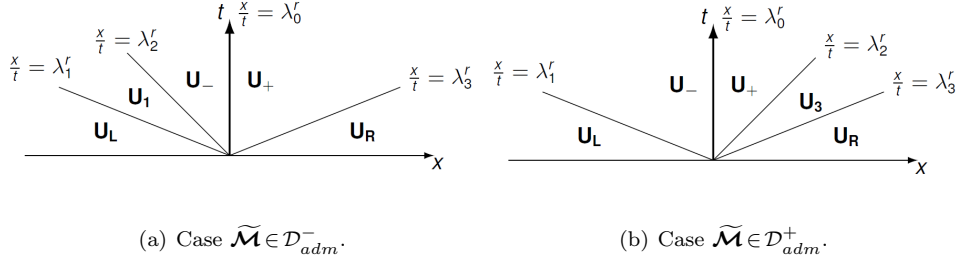


FIG. 3.2. The two possible wave structures of the solution of the Riemann problem (3.16).

Let us introduce the following intermediate states:

$$\begin{aligned} u^* &= \frac{1}{2}(u_L + u_R) - \frac{1}{2a}(\pi_R - \pi_L), & \tau_L^* &= \frac{1}{a}(u^* - \lambda_1^r(\mathbf{U}_L)), & \rho_L^* &= \frac{1}{\tau_L^*}, \\ \pi^* &= \frac{1}{2}(\pi_L + \pi_R) - \frac{a}{2}(u_R - u_L), & \tau_R^* &= \frac{1}{a}(\lambda_3^r(\mathbf{U}_R) - u^*), & \rho_R^* &= \frac{1}{\tau_R^*}, \end{aligned}$$

and define the mass flows $m_{\pm} := \rho_{\pm} u_{\pm}$ through the coupling interface. We also adopt the notation $w_1 = u + \frac{\pi}{a}$ (resp. $w_3 = u - \frac{\pi}{a}$), which is a 1- (resp. 3-) Riemann invariant in the case without measure (see [21]). Then, the following proposition gives the weak entropy self-similar solution to the Riemann problem (3.16).

PROPOSITION 3.1. *Let \mathbf{U}_L and \mathbf{U}_R in Ω^r be two subsonic constant states, and let $\mathcal{M}_{\rho}, \mathcal{M}_{\rho u}, \mathcal{M}_{\rho E}, \mathcal{M}_{\rho \pi}$ be given source terms in $\mathcal{D}_{adm}(\mathbf{U}_L, \mathbf{U}_R)$. Then, the Riemann problem (3.16) has a unique subsonic self-similar solution which we denote by $\mathcal{W}\left(\frac{x}{t}; \mathbf{U}_L, \mathbf{U}_R, \widetilde{\mathcal{M}}\right)$. This function is defined by*

$$\mathcal{W}\left(\frac{x}{t}; \mathbf{U}_L, \mathbf{U}_R, \widetilde{\mathcal{M}}\right) = \begin{cases} \mathbf{U}_L, & \frac{x}{t} < \lambda_1^r(\mathbf{U}_L), \\ \mathbf{U}_1, & \lambda_1^r(\mathbf{U}_L) < \frac{x}{t} < \lambda_2^r(\mathbf{U}_1), \\ \mathbf{U}_-, & \lambda_2^r(\mathbf{U}_1) < \frac{x}{t} < \lambda_0^r, \\ \mathbf{U}_+, & \lambda_0^r < \frac{x}{t} < \lambda_3^r(\mathbf{U}_R), \\ \mathbf{U}_R, & \frac{x}{t} > \lambda_3^r(\mathbf{U}_R), \end{cases}$$

if $\widetilde{\mathcal{M}} \in \mathcal{D}_{adm}^-$, and respectively by

$$\mathcal{W}\left(\frac{x}{t}; \mathbf{U}_L, \mathbf{U}_R, \widetilde{\mathcal{M}}\right) = \begin{cases} \mathbf{U}_L, & \frac{x}{t} < \lambda_1^r(\mathbf{U}_L), \\ \mathbf{U}_-, & \lambda_1^r(\mathbf{U}_L) < \frac{x}{t} < \lambda_0^r, \\ \mathbf{U}_+, & \lambda_0^r < \frac{x}{t} < \lambda_2^r(\mathbf{U}_3), \\ \mathbf{U}_3, & \lambda_2^r(\mathbf{U}_3) < \frac{x}{t} < \lambda_3^r(\mathbf{U}_R), \\ \mathbf{U}_R, & \frac{x}{t} > \lambda_3^r(\mathbf{U}_R), \end{cases}$$

if $\widetilde{\mathcal{M}} \in \mathcal{D}_{adm}^+$. Moreover, in the case where $\widetilde{\mathcal{M}} \in \mathcal{D}_{adm}^-$, we have:

$$\begin{aligned} m_- &= \frac{2au^* + \mathcal{M}_{\rho u} + \mathcal{M}_{\rho\pi}/a - (w_{1L} + 2a\tau_R^*)\mathcal{M}_\rho}{2a\tau_R^*}, & (u, \pi)_1 &= (u, \pi)_-, \tau_1 = \tau_L + \frac{u_1 - u_L}{a}, \\ m_+ &= \frac{2au^* + \mathcal{M}_{\rho u} + \mathcal{M}_{\rho\pi}/a - w_{1L}\mathcal{M}_\rho}{2a\tau_R^*}, & E_1 &= \frac{aE_L + \pi_L u_L - \pi_- u_-}{a}, \\ E_- &= \frac{m_+ E_+ + \pi_+ u_+ - \pi_- u_- - \mathcal{M}_{\rho E}}{m_-}, & E_+ &= \frac{aE_R + \pi_+ u_+ - \pi_R u_R}{a}, \end{aligned}$$

and in the case where $\widetilde{\mathcal{M}} \in \mathcal{D}_{adm}^+$:

$$\begin{aligned} m_- &= \frac{2au^* + \mathcal{M}_{\rho u} - \mathcal{M}_{\rho\pi}/a - w_{3R}\mathcal{M}_\rho}{2a\tau_L^*}, & E_3 &= \frac{aE_R + \pi_+ u_+ - \pi_R u_R}{a}, \\ m_+ &= \frac{2au^* + \mathcal{M}_{\rho u} - \mathcal{M}_{\rho\pi}/a - (w_{3R} - 2a\tau_L^*)\mathcal{M}_\rho}{2a\tau_L^*}, & (u, \pi)_3 &= (u, \pi)_+, \tau_3 = \tau_R + \frac{u_R - u_3}{a}, \\ E_+ &= \frac{m_- E_- + \pi_- u_- - \pi_+ u_+ + \mathcal{M}_{\rho E}}{m_+}, & E_- &= \frac{aE_L + \pi_L u_L - \pi_- u_-}{a}. \end{aligned}$$

Finally, the traces of the solution are given by the following relations:

$$\begin{aligned} u_- &= u^* + \frac{a\mathcal{M}_{\rho u} - \mathcal{M}_{\rho\pi} - (au_R - \pi_R)\mathcal{M}_\rho}{2a(a - m_-)}, & u_+ &= u^* + \frac{a\mathcal{M}_{\rho u} + \mathcal{M}_{\rho\pi} - (\pi_L + au_L)\mathcal{M}_\rho}{2a(a + m_+)}, \\ \pi_- &= \pi^* - \frac{a\mathcal{M}_{\rho u} - \mathcal{M}_{\rho\pi} + (\pi_R - au_R)\mathcal{M}_\rho}{2(a - m_-)}, & \pi_+ &= \pi^* + \frac{a\mathcal{M}_{\rho u} + \mathcal{M}_{\rho\pi} - (\pi_L + au_L)\mathcal{M}_\rho}{2(a + m_+)}, \end{aligned}$$

together with $\rho_- = \frac{m_-}{u_-}$, $\rho_+ = \frac{m_+}{u_+}$.

Proof. We only consider the case $\widetilde{\mathcal{M}} \in \mathcal{D}_{adm}^-$, the case $\widetilde{\mathcal{M}} \in \mathcal{D}_{adm}^+$ being symmetric. Recall that all characteristic fields are linearly degenerate so that only contact discontinuities are involved.

The continuity of the wave speeds for the λ_1^r , λ_2^r and λ_3^r - waves reads:

$$\tau_1 = \tau_L + \frac{u_1 - u_L}{a}, \quad (3.19a)$$

$$u_1 = u_-, \quad (3.19b)$$

$$u_R + \frac{a}{\rho_R} = u_+ + \frac{a}{\rho_+}. \quad (3.19c)$$

Through the steady interface $\lambda_0^r = 0$, the relations (3.3) read

$$\mathcal{M}_\rho = m_+ - m_-, \quad (3.20a)$$

$$\mathcal{M}_{\rho u} = m_+ u_+ + \pi_+ - m_- u_- - \pi_-, \quad (3.20b)$$

$$\mathcal{M}_{\rho E} = m_+ E_+ + \pi_+ u_+ - m_- E_- - \pi_- u_-, \quad (3.20c)$$

$$\mathcal{M}_{\rho\pi} = m_+ \pi_+ + a^2 u_+ - m_- \pi_- - a^2 u_-. \quad (3.20d)$$

Across the λ_1^r -wave, the Rankine-Hugoniot relations on the momentum and energy balance equations read

$$\begin{cases} \lambda_1^r(\mathbf{U}_L)(\rho_1 u_1 - \rho_L u_L) = \rho_1 u_1^2 + \pi_1 - \rho_L u_L^2 - \pi_L, \\ \lambda_1^r(\mathbf{U}_L)(\rho_1 E_1 - \rho_L E_L) = \rho_1 E_1 u_1 + \pi_1 u_1 - \rho_L E_L u_L - \pi_L u_L, \end{cases}$$

which implies

$$\pi_1 - \pi_L - au_L + au_1 = 0, \quad (3.21a)$$

$$E_1 - E_L - \frac{1}{a}(\pi_L u_L - \pi_1 u_1) = 0. \quad (3.21b)$$

Considering now the λ_2^r -wave and the Rankine-Hugoniot relations applied to the energy balance equation, we have:

$$\lambda_2^r(\mathbf{U}_1)(\rho_- E_- - \rho_1 E_1) = \rho_- E_- u_- + \pi_- u_- - \rho_1 E_1 u_1 - \pi_1 u_1,$$

and using (3.19b), we obtain:

$$\pi_1 = \pi_-. \quad (3.22)$$

From (3.19b) and (3.22), we obtain that the equation (3.21a) is equivalent to:

$$\pi_- - \pi_L - au_L + au_- = 0, \quad (3.23)$$

which means $w_{1L} = w_{1-}$. Next, through the λ_3^r -contact, the Rankine-Hugoniot relations on the momentum and energy balance equations read

$$\begin{cases} \lambda_3^r(\mathbf{U}_+)(\rho_R u_R - \rho_+ u_+) = \rho_R u_R^2 + \pi_R - \rho_+ u_+^2 - \pi_+, \\ \lambda_3^r(\mathbf{U}_+)(\rho_R E_R - \rho_+ E_+) = \rho_R E_R u_R + \pi_R u_R - \rho_+ E_+ u_+ - \pi_+ u_+, \end{cases}$$

and it follows:

$$au_R - au_+ = \pi_R - \pi_+, \text{ or } w_{3+} = w_{3R}, \quad (3.24a)$$

$$aE_R - aE_+ = \pi_R u_R - \pi_+ u_+. \quad (3.24b)$$

The traces u_-, u_+, π_-, π_+ are then obtained by solving the linear system of equations (3.20b), (3.20d), (3.23) and (3.24a), which is left to the reader.

From relations (3.24b) and (3.20c), we get:

$$E_+ = \frac{aE_R + \pi_+ u_+ - \pi_R u_R}{a}, \quad (3.25)$$

and

$$E_- = \frac{m_+ E_+ + \pi_+ u_+ - \pi_- u_- - \mathcal{M}_{\rho E}}{m_-}. \quad (3.26)$$

Next, we determine the expressions of m_+ and m_- . The expression of the state u_+ gives

$$2a(a + m_+)(u_+ - u^*) = a\mathcal{M}_{\rho u} + \mathcal{M}_{\rho\pi} - (au^* + \pi^*)\mathcal{M}_{\rho}, \quad (3.27)$$

then, multiplying the relation (3.19c) by ρ_+ , we have

$$m_+ + a = \rho_+ \left(u_R + \frac{a}{\rho_R} \right),$$

so that

$$(a + m_+)u_+ = m_+ \left(u_R + \frac{a}{\rho_R} \right),$$

and

$$\begin{aligned} 2a(a+m_+)(u_+ - u^*) &= 2am_+(u_R + \frac{a}{\rho_R}) - 2a^2u^* - 2am_+u^* \\ &= \frac{2a^2}{\rho_R^*}m_+ - 2a^2u^*. \end{aligned}$$

Moreover, from (3.27), we obtain:

$$m_+ = \frac{2a^2u^* + a\mathcal{M}_{\rho u} + \mathcal{M}_{\rho\pi} - (\pi_L + au_L)\mathcal{M}_\rho}{2a^2\tau_R^*}, \quad (3.28)$$

and from (3.20a) we finally get

$$m_- = \frac{2a^2u^* + a\mathcal{M}_{\rho u} + \mathcal{M}_{\rho\pi} - (\pi_L + au_L + 2a^2\tau_R^*)\mathcal{M}_\rho}{2a^2\tau_R^*}, \quad (3.29)$$

which ends the proof of the case under consideration. \square

We check that we do recover the formulas of [5] in the case $\mathcal{M}_\rho = 0$. Now, we can define the numerical fluxes (3.6)(3.7) introduced in §3.3. Recall that we have noted Π the projection operator defined as $\Pi: (x_1, x_2, x_3, x_4) \in \mathbb{R}^4 \mapsto (x_1, x_2, x_3) \in \mathbb{R}^3$. Then, the numerical fluxes $(\mathbf{g}_L)_j^n$ and $(\mathbf{g}_R)_j^n$ are given by the following Godunov type numerical fluxes:

$$\begin{aligned} (\mathbf{g}_L)_j^n &= \Pi \mathbf{F} \left(\mathcal{W} \left(0^-; \mathbf{U}_{j-\frac{1}{2}}^{(n)}, \mathbf{U}_{j+\frac{1}{2}}^{(n)}, \widetilde{\mathcal{M}}_j^{(n)} \right) \right), \quad j \leq 0, \\ (\mathbf{g}_R)_j^n &= \Pi \mathbf{F} \left(\mathcal{W} \left(0^+; \mathbf{U}_{j-\frac{1}{2}}^{(n)}, \mathbf{U}_{j+\frac{1}{2}}^{(n)}, \widetilde{\mathcal{M}}_j^{(n)} \right) \right), \quad j \geq 0, \end{aligned}$$

where the weights $\widetilde{\mathcal{M}}_j^{(n)}$ are defined by

$$\widetilde{\mathcal{M}}_j^{(n)} = \begin{cases} \mathbf{0}, & \text{if } j \neq 0, \\ \mathcal{M}^{(n)}, & \text{if } j = 0. \end{cases}$$

Note that for $j \neq 0$, the solution \mathcal{W} of the Riemann problem may be discontinuous at 0 but the flux \mathbf{F} is continuous (see indeed (3.3)) so that the choice to take the value of $\mathcal{W}(0^\pm)$ is not important; the numerical flux is just the projection of the exact Godunov flux for the relaxation system. While, for $j = 0$, there are two fluxes linked by the discrete coupling condition (discrete version of (2.7))

$$(\mathbf{g}_R)_0^n - (\mathbf{g}_L)_0^n = \mathcal{M}^{(n)}.$$

This situation with two fluxes at an interface is now classical for finite volume schemes approximating a system with geometric source term.

3.7. Preserving equilibria

Next, it remains to determine the additional weight $\mathcal{M}_{\rho\pi}$ such that the relaxation approach preserves the equilibrium solutions of the original coupled problem (2.5) (see definition 3.1). The following proposition gives this weight.

PROPOSITION 3.2. *Let \mathbf{u}_L and \mathbf{u}_R be two subsonic constant states belonging to Ω and a weight $\mathcal{M} = (\mathcal{M}_\rho, \mathcal{M}_{\rho u}, \mathcal{M}_{\rho E})^T$ such that the relation (3.15) holds true. Let us*

define the mass flows through the interface $m_{\pm}^e = m_{\pm}^e(\mathbf{u}_L, \mathbf{u}_R, \mathcal{M}_\rho, \mathcal{M}_{\rho u})$ as

$$m_+^e := \frac{\mathcal{M}_{\rho u} + 2au^* + (a\tau_L^* - u^*)\mathcal{M}_\rho}{a(\tau_L^* + \tau_R^*)}, \quad (3.30)$$

$$m_-^e := m_+^e - \mathcal{M}_\rho, \quad (3.31)$$

together with the weight $\mathcal{M}_{\rho\pi}^e = \mathcal{M}_{\rho\pi}^e(\mathbf{u}_L, \mathbf{u}_R, \mathcal{M}_\rho, \mathcal{M}_{\rho u})$, given by

$$\mathcal{M}_{\rho\pi}^e := \mathcal{P}_R m_+^e - \mathcal{P}_L m_-^e, \quad (3.32)$$

where $\mathcal{P}_\alpha = p_\alpha(\tau_\alpha, \varepsilon_\alpha) + a^2\tau_\alpha$, with $\alpha = L, R$. Then, the relaxation method in Proposition 3.1 with the weight $\widetilde{\mathcal{M}}^e = (\mathcal{M}_\rho, \mathcal{M}_{\rho u}, \mathcal{M}_{\rho E}, \mathcal{M}_{\rho\pi}^e)^T$ preserves the equilibrium solutions of the coupled problem (2.5).

Proof. Let $\mathbf{u}_L, \mathbf{u}_R$ in Ω be such that the relation (3.15) holds true and assume that $m_+^e, m_-^e, \mathcal{P}_R, \mathcal{P}_L$, and $\mathcal{M}_{\rho\pi}^e$ satisfy relations (3.30) to (3.32). Let us show that

$$\mathcal{W}\left(\frac{x}{t}; \mathbf{U}_L, \mathbf{U}_R, \widetilde{\mathcal{M}}^e\right) = \begin{cases} \mathbf{U}_L, & x < 0, \\ \mathbf{U}_R, & x > 0, \end{cases} \quad (3.33)$$

where $\mathbf{U}_\alpha = (\mathbf{u}_\alpha, \rho_\alpha p_\alpha)$, $\alpha = L, R$ and where \mathcal{W} is the solution to the Riemann problem (3.16) solved explicitly in Proposition 3.1. Therefore, we must prove that

$$\begin{cases} \mathbf{U}_- = \mathbf{U}_1 = \mathbf{U}_L & \text{and} & \mathbf{U}_+ = \mathbf{U}_R & \text{if} & \widetilde{\mathcal{M}}^e \in \mathcal{D}_{adm}^-, \\ \mathbf{U}_+ = \mathbf{U}_3 = \mathbf{U}_R & \text{and} & \mathbf{U}_- = \mathbf{U}_L & \text{if} & \widetilde{\mathcal{M}}^e \in \mathcal{D}_{adm}^+. \end{cases}$$

Here we present only the evaluation of the trace state \mathbf{U}_+ since the approach is similar for the other states.

By Proposition 3.1, note that the mass flows m_{\pm}^e are identical for the two considered wave structures. Let $m_\alpha = \rho_\alpha u_\alpha$, $\alpha = L, R$. From (3.15), we have:

$$\mathcal{M}_\rho = m_R - m_L, \quad (3.34a)$$

$$\mathcal{M}_{\rho u} = m_R u_R + p_R - m_L u_L - p_L, \quad (3.34b)$$

$$\mathcal{M}_{\rho E} = m_R E_R + p_R u_R - m_L E_L - p_L u_L. \quad (3.34c)$$

By plugging the weights \mathcal{M}_ρ and $\mathcal{M}_{\rho u}$ into (3.30), and using the relations $a\tau_L^* - u^* = a\tau_L - u_L$ and $\pi_\alpha = p_\alpha$ with $\alpha = L, R$, we obtain

$$m_+^e = \frac{m_R(a(\tau_L + \tau_R) + u_R - u_L)}{a(\tau_L^* + \tau_R^*)}.$$

From the definition of the intermediate states τ_L^* and τ_R^* , we have:

$$a(\tau_L^* + \tau_R^*) = \lambda_3(\mathbf{U}_R) - \lambda_1(\mathbf{U}_L) = a(\tau_L + \tau_R) + u_R - u_L,$$

and

$$m_+^e = m_R. \quad (3.35)$$

Using relation (3.34a), it follows that

$$m_-^e = m_+^e - \mathcal{M}_\rho = m_R - (m_R - m_L) = m_L. \quad (3.36)$$

Let us consider now the state u_+ . From Proposition 3.1, the equalities $\pi_\alpha = p_\alpha$ with $\alpha = L, R$, the definitions of the intermediate state u^* and the weight $\mathcal{M}_{\rho\pi}^e$ (3.32), together with the relations (3.34a), (3.34b), (3.35) and (3.36), we get

$$\begin{aligned}
 u_+ &= u^* + \frac{a\mathcal{M}_{\rho u} + \mathcal{M}_{\rho\pi}^e - (\pi_L + au_L)\mathcal{M}_\rho}{2a(a+m_+^e)} \\
 &= \frac{u_R + u_L}{2} + \frac{p_L - p_R}{2a} + \frac{a(m_R u_R + p_R - m_L u_L - p_L) + (p_R + a^2 \tau_R)m_R}{2a(a+m_R)} \\
 &\quad - \frac{(p_L + a^2 \tau_L)m_L}{2a(a+m_R)} - \frac{(p_L + au_L)(m_R - m_L)}{2a(a+m_R)} \\
 &= \frac{u_R + u_L}{2} + \frac{p_L - p_R}{2a} + \frac{au_R(m_R + a) + p_R(a + m_R) - (p_L + au_L)(a + m_R)}{2a(a+m_R)} \\
 &= u_R,
 \end{aligned}$$

and it follows immediately that $\rho_+ = m_+^e/u_+ = m_R/u_R = \rho_R$. Next, using similar arguments together with the definition of the state π^* , we obtain

$$\begin{aligned}
 \pi_+ &= \pi^* + \frac{a\mathcal{M}_{\rho u} + \mathcal{M}_{\rho\pi}^e - (\pi_L + au_L)\mathcal{M}_\rho}{2(a+m_+^e)} \\
 &= \frac{\pi_L + \pi_R}{2} + \frac{a(u_L - u_R)}{2} + \frac{(a+m_R)(au_R + p_R - p_L - au_L)}{2(a+m_R)} \\
 &= \frac{p_L + p_R}{2} + \frac{a(u_L - u_R)}{2} + \frac{p_R - p_L + a(u_R - u_L)}{2} \\
 &= p_R.
 \end{aligned}$$

Now, for the evaluation of E_+ , two cases must be considered. If $\widetilde{\mathcal{M}}^e \in \mathcal{D}_{adm}^-$, we use the Proposition 3.1 and the equality $\pi_+ u_+ = \pi_R u_R$, and immediately get

$$E_+ = \frac{aE_R + \pi_+ u_+ - \pi_R u_R}{a} = E_R.$$

On the other hand, if $\widetilde{\mathcal{M}}^e \in \mathcal{D}_{adm}^+$, the energy is given by:

$$E_+ = \frac{m_- E_- + \pi_- u_- - \pi_+ u_+ + \mathcal{M}_\rho E}{m_+}.$$

Moreover, a computation similar to those made for the states u_+ and π_+ gives $\pi_- u_- = \pi_L u_L$ so that $E_- = E_L$ (cf. Proposition 3.1). Using these two last relations with equation (3.34c), we finally obtain

$$E_+ = \frac{m_L E_L + p_L u_L - p_R u_R + m_R E_R + p_R u_R - m_L E_L - p_L u_L}{m_R} = E_R,$$

which ends the proof. \square

3.8. Choices of the measure source terms

The measure source terms can be determined in a dynamic way, at each time step of a discrete time algorithm. We present different choices according to the chosen objective.

3.8.1. Flux coupling and state coupling. In model adaptation, one couples two different models at an interface which is fixed during a time interval of length the time step, but may vary at each time step. The position of the interface is optimized in such a way that in the region where one computes the coarse model, the model error between the fine and coarse models (evaluated by some indicator) is less than some given tolerance. Then, the coupling condition should not have a great impact, and one chooses at each time step the same, either a conservative (flux coupling) or a non conservative (state coupling) approach [56].

At time $t^{(n)}$, the coupling condition (2.7) at the discrete level reads

$$\mathcal{M}^{(n)} = (\mathbf{g}_R)_0^n - (\mathbf{g}_L)_0^n. \quad (3.37)$$

The choice of a flux coupling leads to $\mathcal{M}^{(n)} = \mathbf{0}$ for all n in \mathbb{N} . In order to ensure the state coupling we use an approach based on the introduction of reconstructed states. This procedure was proposed in [6], formalized in [8] and then adapted in the context of coupling with source terms [34]. The method allows to impose continuity (at the coupling interface) of a given set of variables. This can be related to other non-conservative approaches as the single fluid algorithm introduced in [1] and also to well balanced schemes in the spirit of [48, 44], see [47].

The numerical fluxes $(\mathbf{g}_L)_0^n$ and $(\mathbf{g}_R)_0^n$ are given by:

$$(\mathbf{g}_L)_0^n = \mathbf{g}_L \left(\mathbf{u}_{-\frac{1}{2}}^{(n)}, \bar{\mathbf{u}}_{+\frac{1}{2}}^{(n)} \right), \quad (\mathbf{g}_R)_0^n = \mathbf{g}_R \left(\bar{\mathbf{u}}_{-\frac{1}{2}}^{(n)}, \mathbf{u}_{+\frac{1}{2}}^{(n)} \right),$$

where (using the notations introduced in section 2.2), the reconstructed states $\bar{\mathbf{u}}_{-\frac{1}{2}}^{(n)}$ and $\bar{\mathbf{u}}_{+\frac{1}{2}}^{(n)}$ are defined as

$$\bar{\mathbf{u}}_{-\frac{1}{2}}^{(n)} := \varphi_R^{-1} \circ \varphi_L \left(\mathbf{u}_{-\frac{1}{2}}^{(n)} \right), \quad \bar{\mathbf{u}}_{+\frac{1}{2}}^{(n)} := \varphi_L^{-1} \circ \varphi_R \left(\mathbf{u}_{+\frac{1}{2}}^{(n)} \right).$$

We refer the reader to [6, 34] for examples of changes of variables.

3.8.2. Dynamic evaluation: constrained optimization problems. The aim of this section is the evaluation of an optimal source term relative to some cost function. At each time step, given a convex cost function \mathcal{J} and a non-empty convex subset $\mathcal{D} \subset \mathcal{D}_{adm}$, the problem consists in finding an optimal mass $\widetilde{\mathcal{M}}_{opt}^e$ in \mathcal{D} such that

$$\mathcal{J}(\widetilde{\mathcal{M}}_{opt}^e) \leq \mathcal{J}(\widetilde{\mathcal{M}}^e), \quad \forall \widetilde{\mathcal{M}}^e \in \mathcal{D}. \quad (3.38)$$

The objective function and the space of constraints are defined by physical motivations, for instance in order to impose a mass flow on one side of the interface, or else to impose a cooling; the cost function is then constructed in relation with the objective (see §4.2 for several examples). At each time step, when this optimization problem is solved, we set $\mathcal{M}^{(n)} = \Pi \widetilde{\mathcal{M}}_{opt}^e$, where Π is the projection operator introduced in §3.6.

4. Numerical experiments.

We use a computational domain $(-0.5, 0.5)$ with 200 elements. The coupling interface is at $x=0$. Moreover, the final time of the simulation is $T=0.12$ and the ratios of specific heats are $\gamma_L=1.4$ and $\gamma_R=1.28$. In order to preserve equilibrium solutions of the problem, we impose $\mathcal{M}_{\rho\pi} = \mathcal{M}_{\rho\pi}^e$, where $\mathcal{M}_{\rho\pi}^e$ is defined by (3.32). The following test cases are based on Riemann problems where the discontinuity

is located at $x=0$ and the different initial data are given in table 4.1. Numerical experiments using state coupling are based on the continuity of primitive variables $\varphi_\alpha(\mathbf{u}) = (\rho, u, p_\alpha)$, $\alpha = L, R$.

Test case	ρ_L	u_L	p_L	ρ_R	u_R	p_R
1	1.6	0.4	2.35	1.6	0.4	2.35
2,3,4,5	1.6	0.4	2.35	1.4	0.4	1.9

TABLE 4.1. Initial conditions for different test cases.

4.1. Flux coupling vs. state coupling

Test case 1. In this first test case, we consider uniform initial data given in table 4.1. The density, velocity, pressure and internal energy profiles are displayed in figure 4.1 when applying flux coupling and state coupling in primitive variables (ρ, u, p) . Figure 4.2 displays the relative conservation losses in mass, momentum and total energy, obtained by integrating the conservation laws over space and time.

Results show that the uniform profile is not respected for the flux coupling. Nevertheless, the method is strictly conservative with zero conservation losses. In contrast, the uniform profiles of ρ , u and p are preserved when using state coupling as expected. Note also that the total energy is not conserved [10, 34]. Finally, we remark that the internal energy ε is necessarily discontinuous at the coupling interface due to the different pressure laws (see equation (2.2)).

Test case 2. In the second test case, we still compare the various profiles based on conservative coupling or state coupling in primitive variables for a shock tube problem (see table 4.1). The density, velocity, pressure, internal energy and mass flow rates are shown in figure 4.3, while figure 4.4 displays the relative conservation losses.

Results are similar to test case 1: the flux coupling method is strictly conservative and the state coupling method ensures continuity of primitive variables through the coupling interface.

4.2. Optimal source terms

Hereafter, $m_{ref} \in \mathbb{R}^*$ and $\varepsilon_{ref} > 0$ will denote target values. At each time $t^{(n)}$, $n \in \mathbb{N}$, we consider the discrete states $\mathbf{U}_L = \mathbf{U}_{-\frac{1}{2}}^{(n)}$, and $\mathbf{U}_R = \mathbf{U}_{+\frac{1}{2}}^{(n)}$ at the coupling interface to define objective functions.

For the numerical resolution of constrained optimization problems (3.38), we use the gradient projection algorithm defined by the sequence

$$\begin{cases} \widetilde{\mathcal{M}}_0^e & \text{given} \\ \widetilde{\mathcal{M}}_{k+1}^e = \mathcal{P}_{\mathcal{D}} \left(\widetilde{\mathcal{M}}_k^e - \mu_{\mathcal{J}} \nabla \mathcal{J}(\widetilde{\mathcal{M}}_k^e) \right), & k \in \mathbb{N}, \end{cases} \quad (4.1)$$

where $\mathcal{P}_{\mathcal{D}}$ is the projection operator onto the set of constraints \mathcal{D} . In our numerical examples, $\mathcal{P}_{\mathcal{D}}$ is known explicitly and thus motivates the use of this algorithm. For a given tolerance $\eta > 0$, the stopping condition of the algorithm will be $\|\widetilde{\mathcal{M}}_{k+1}^e - \widetilde{\mathcal{M}}_k^e\| < \eta \|\widetilde{\mathcal{M}}_0^e\|$, where $\|\cdot\|$ denotes the usual Euclidean norm and $\eta = 10^{-12}$.

Test case 3. In this test case, we focus on an optimization problem that imposes the mass flow at the right hand side of the coupling interface, while controlling the conservation losses in mass and momentum. The applications we have in mind

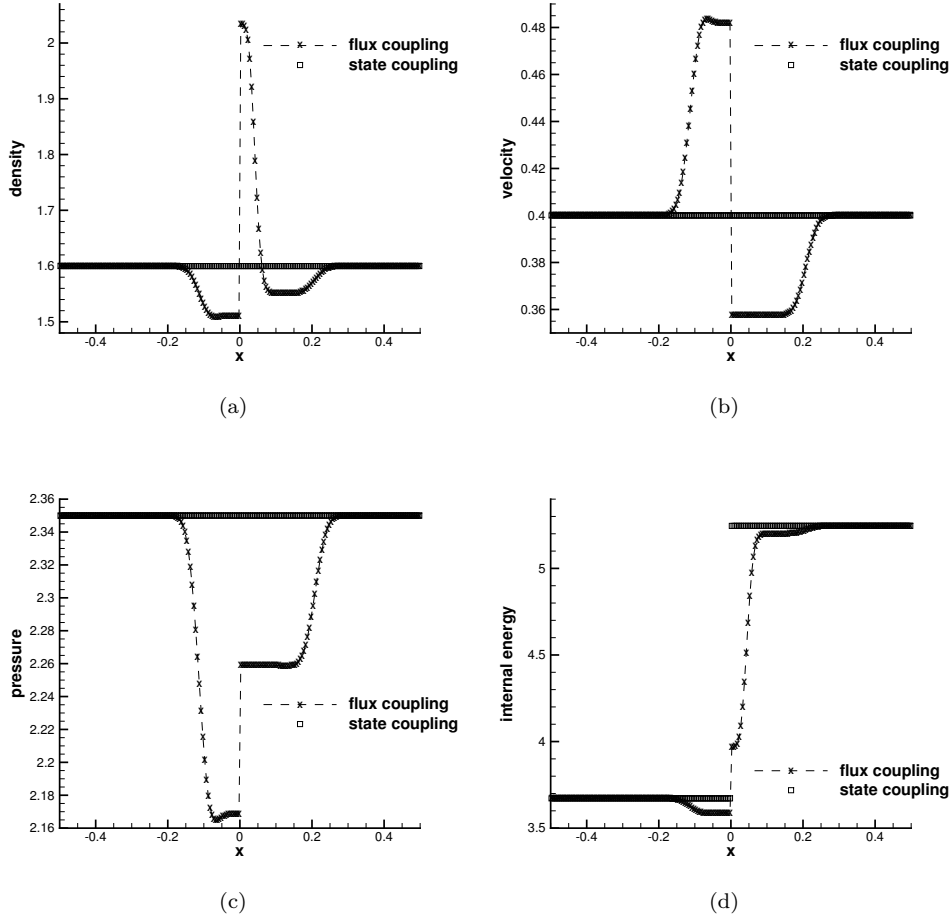


FIG. 4.1. *Conservative coupling vs. state coupling for test case 1: profiles of primitive variables (ρ, u, p) and internal energy.*

here concern unsteady flow control techniques. Among the most widely used techniques, one can quote the experimental methods of fluid blowing or suction which have a strong influence on the flow features such as lift improvement, drag reduction, vortex breakdown delay, flow separation control [33, 32, 57]. In the experimental studies, a typical objective may be to optimize the blowing position for maximizing control and minimizing the blowing mass flow rate. A simplified representation of these flow control problems would consider the minimization of the objective function $(\mathcal{M}_\rho, \mathcal{M}_{\rho u}) \in \mathbb{R}^2 \mapsto \mathcal{J}_1(\mathcal{M}_\rho, \mathcal{M}_{\rho u}) \in \mathbb{R}$ defined by

$$\mathcal{J}_1(\mathcal{M}_\rho, \mathcal{M}_{\rho u}) := \left(\frac{m_+^e(\mathbf{u}_L, \mathbf{u}_R, \mathcal{M}_\rho, \mathcal{M}_{\rho u}) - m_{ref}}{m_{ref}} \right)^2 + \kappa^2 \left[\left(\frac{\mathcal{M}_{\rho u}}{a^2 \tau_R} \right)^2 + \left(\frac{\mathcal{M}_\rho}{a} \right)^2 \right], \quad (4.2)$$

over the set $\mathcal{D}_1 = [-0.5, 0.5]^2$, \mathcal{J}_1 is strictly convex (see the Appendix A). We further assume $\mathcal{M}_{\rho E}^{(n)} = 0$, for all n in \mathbb{N} . The parameter $\kappa > 0$ quantifies the cost of the control.

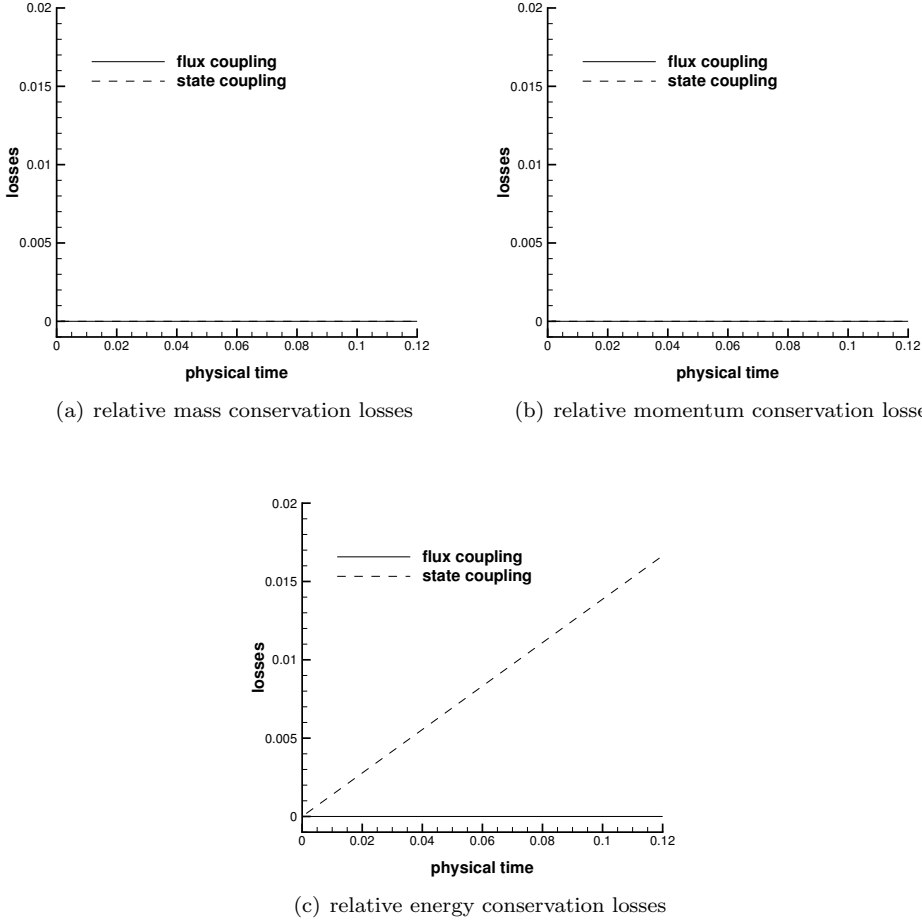


FIG. 4.2. *Relative conservation losses for test case 1.*

Minimizing \mathcal{J}_1 over \mathbb{R}^2 with $\kappa=0$ imposes $m_+^e = m_{ref}$, while conservation losses \mathcal{M}_ρ and $\mathcal{M}_{\rho u}$ are set to zero when κ tends to infinity. In this example, we take $m_{ref} = 0.75$.

Figure 4.5 represents the density, velocity, pressure, internal energy and mass flow profiles for three values of the control parameter: $\kappa = 0.01$, $\kappa = 1$ and $\kappa = 100$ with respective parameter value $\mu_{\mathcal{J}} = 0.1$, 10^{-2} and 10^{-4} in algorithm (4.1). These parameters were chosen experimentally to obtain a good convergence of the algorithm. Figure 4.6 displays the relative conservation losses in mass, momentum and total energy.

The total energy is conserved for all values of κ as expected. Comparing figures 4.3 and 4.5, we observe that the profiles of density, velocity, pressure, internal energy and mass flow obtained from the minimization of the cost function \mathcal{J}_1 with $\kappa = 100$ are similar to those obtained by the flux coupling method. In contrast, the right trace of the mass flow tends to the value imposed $m_{ref} = 0.75$ when κ tends to zero. Finally, we observe that the amplitudes of the conservation losses in mass and momentum increase when the control parameter κ decreases.

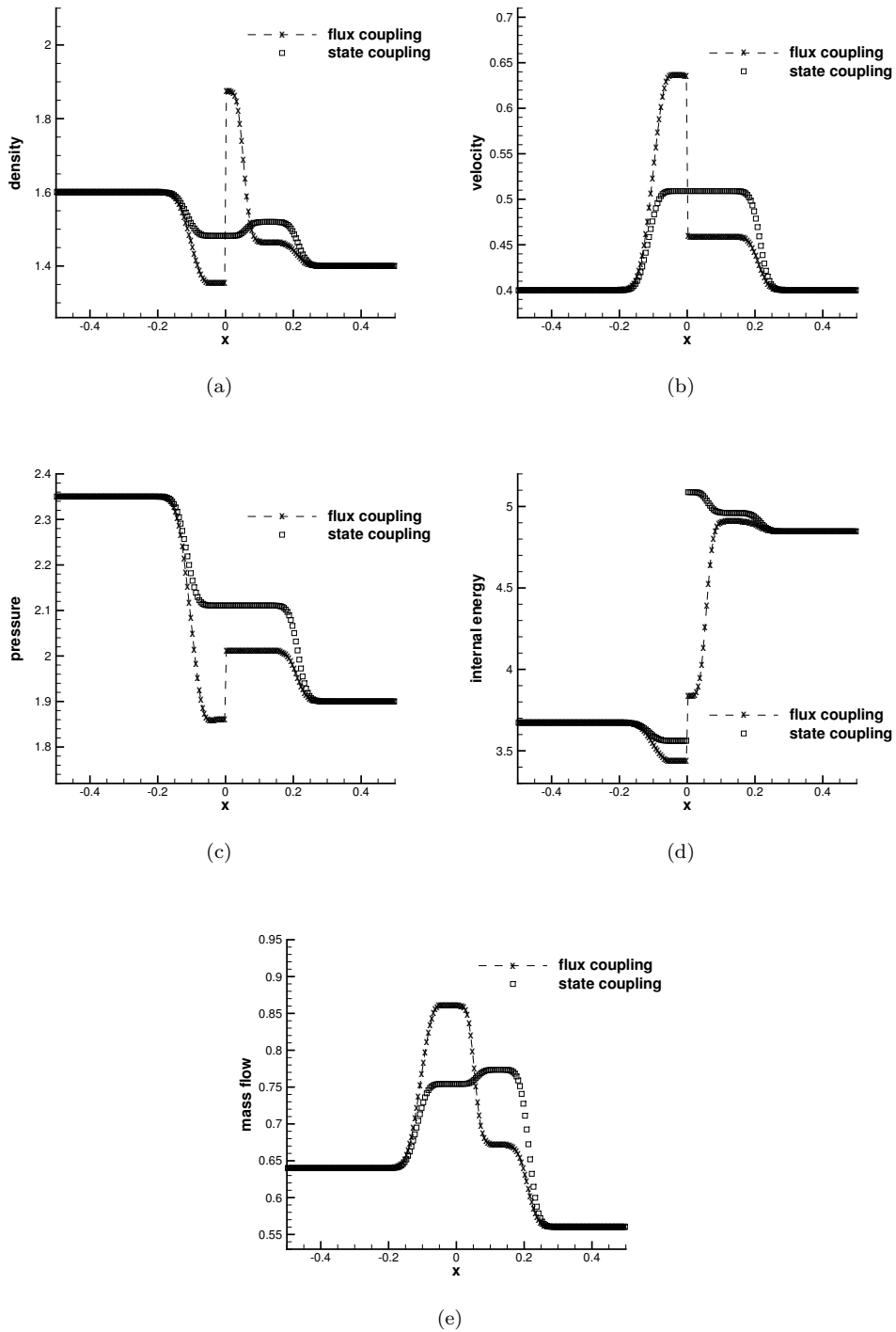
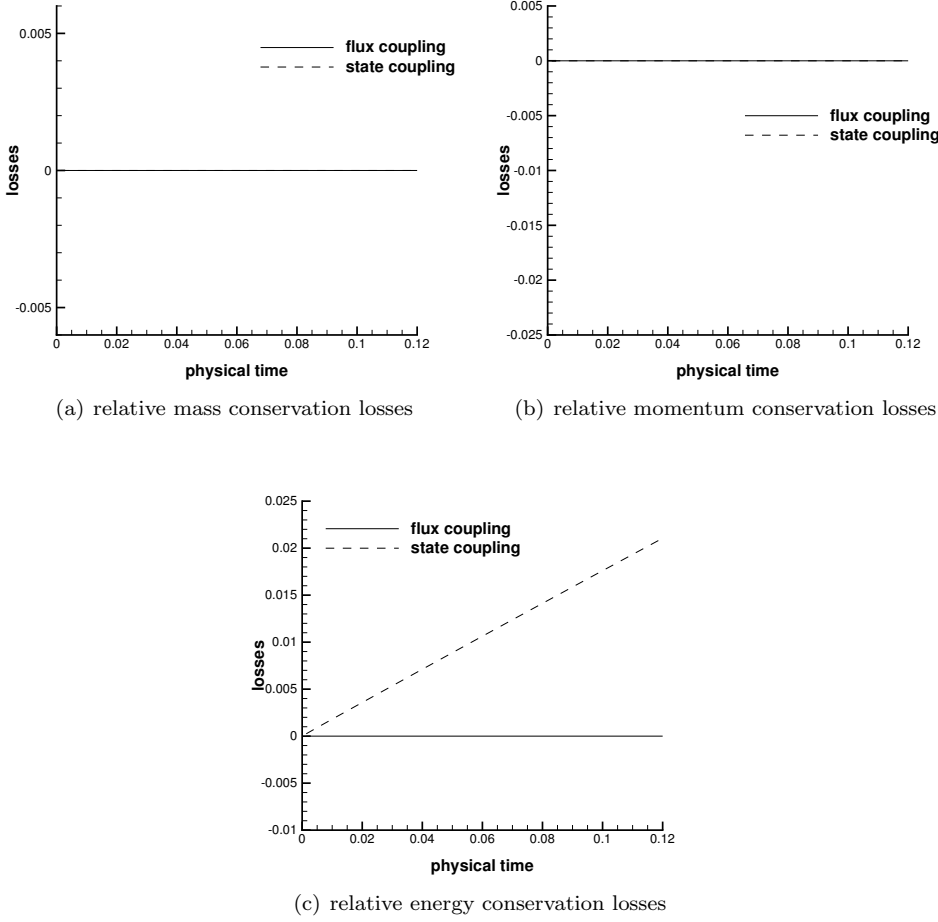


FIG. 4.3. Conservative coupling vs. state coupling for test case 2: profiles of primitive variables (ρ, u, p), mass flow and internal energy.


 FIG. 4.4. *Relative conservation losses for test case 2.*

Test case 4. Now, for this test case, we set $\mathcal{M}_\rho^{(n)} = \mathcal{M}_{\rho u}^{(n)} = 0$ at each time step and try to minimize the function $\mathcal{M}_{\rho E} \in \mathbb{R} \mapsto \mathcal{J}_2(\mathcal{M}_{\rho E}) \in \mathbb{R}$ defined by

$$\mathcal{J}_2(\mathcal{M}_{\rho E}) := \left(\frac{\varepsilon_+ - \varepsilon_{ref}}{\varepsilon_{ref}} \right)^2 + \kappa^2 \left(\frac{\mathcal{M}_{\rho E}}{a^3 \tau_R^2} \right)^2, \quad (4.3)$$

with $\varepsilon_{ref} = 3.5$ over the set $\mathcal{D}_2 = [-1.5, 1.5]$. This function is a strictly convex function of its argument (see the Appendix A). It allows to impose the temperature through the value of ε_+ and thus imposes cooling ($\mathcal{M}_{\rho E} < 0$) or heating ($\mathcal{M}_{\rho E} > 0$) at the right of the interface $x = 0$, while controlling the conservative losses. Film cooling constitutes an example of such minimization problem. The efficiency of gas turbines may be improved and fuel consumption reduced with higher combustor exit flow temperatures. However, higher temperatures may induce turbine components damages. To insure their thermal protection, a protective coat of cool air separates the turbine wall from the hot gases [31]. The boundary between the film and the gases can be modelled as an interface through which the intake air mass and energy losses must be considered.

In figure 4.7, we present the density, velocity, pressure, internal energy and mass flow profiles for different values of the control parameter. Figure 4.8 displays the relative conservation losses for these values of κ . Here again, the conclusions are similar to those of the previous test case.

Test case 5. Finally, in this last test case, we assume that $\mathcal{M}_{\rho u}^{(n)} = 0$ for all n in \mathbb{N} and minimize the function $(\mathcal{M}_\rho, \mathcal{M}_{\rho E}) \in \mathbb{R}^2 \mapsto \mathcal{J}_3(\mathcal{M}_\rho, \mathcal{M}_{\rho E}) \in \mathbb{R}$ defined by

$$\mathcal{J}_3(\mathcal{M}_\rho, \mathcal{M}_{\rho E}) := \left(\frac{\varepsilon_+ - \varepsilon_{ref}}{\varepsilon_{ref}} \right)^2 + \kappa^2 \left[\left(\frac{\mathcal{M}_\rho}{a} \right)^2 + \left(\frac{\mathcal{M}_{\rho E}}{a^3 \tau_R^2} \right)^2 \right],$$

over the set $\mathcal{D}_3 = [0, 0.5] \times [-0.5, 0]$. We set $\varepsilon_{ref} = 3.5$, this function imposes the temperature while controlling the conservative losses. The main difference consists in film cooling by injection of mass ($\mathcal{M}_\rho > 0$) and loss of energy ($\mathcal{M}_{\rho E} < 0$). Figure 4.9 presents the density, velocity, pressure, internal energy and mass flow profiles as a function of the control parameter. Figure 4.10 displays the relative conservation losses for these three values of κ . Here again, the conclusions are similar to the other optimization problems.

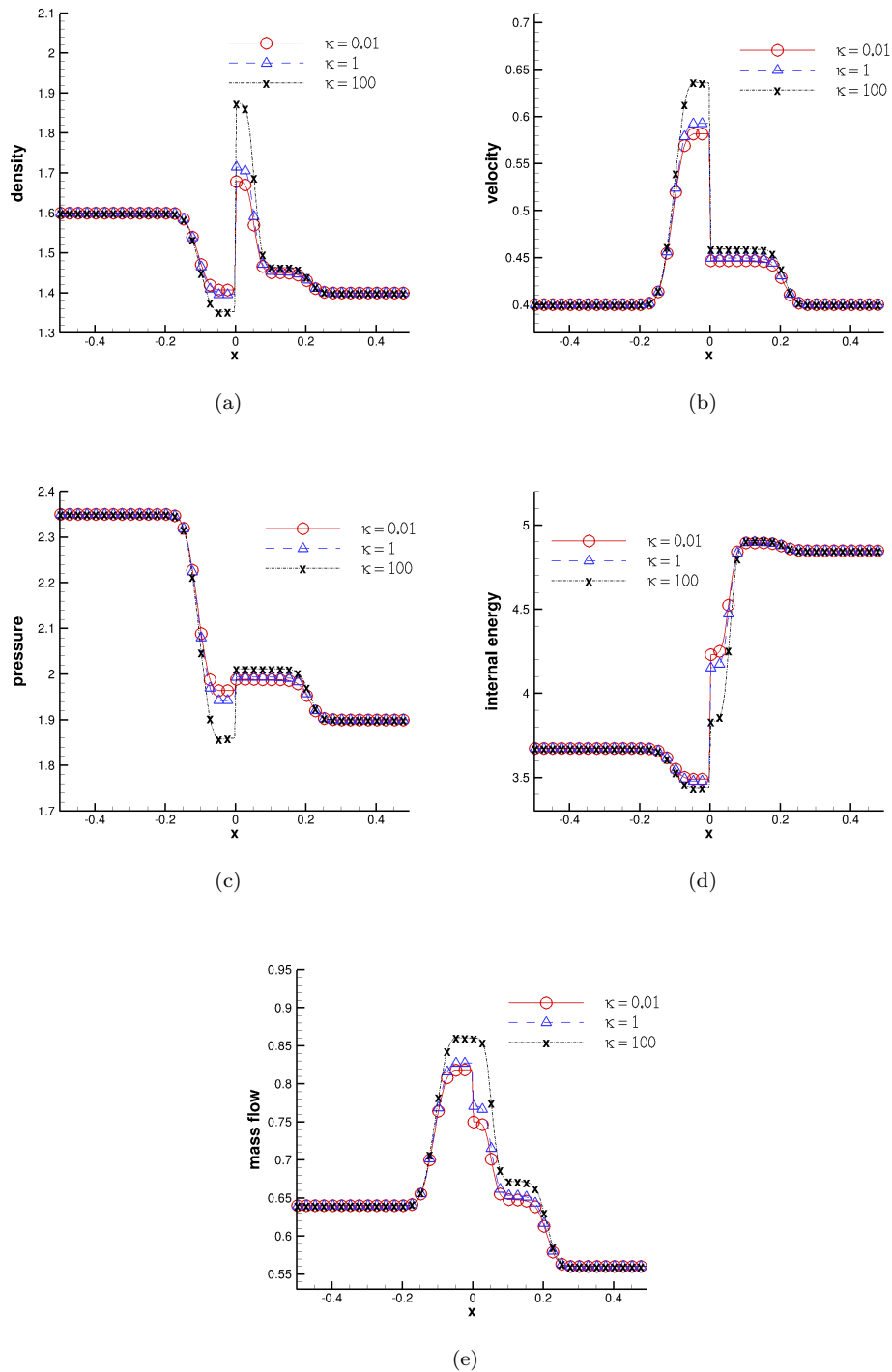
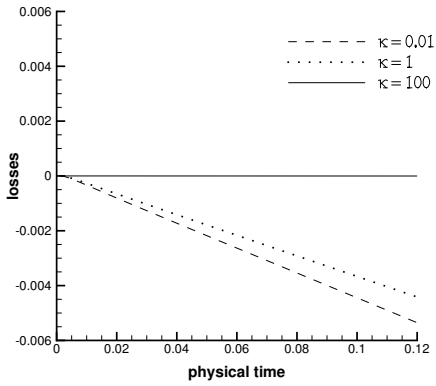
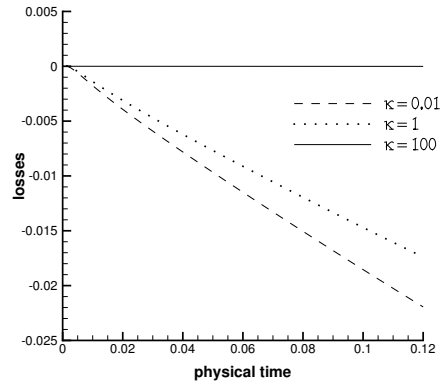


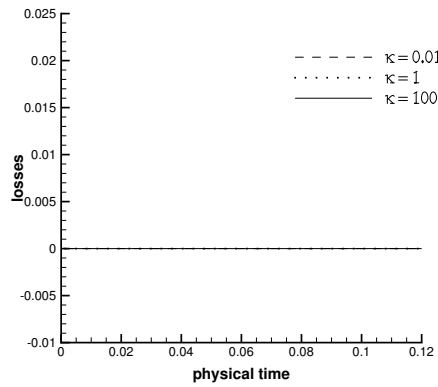
FIG. 4.5. Constrained optimization for test case 3: profiles of primitive variables (ρ, u, p), mass flow and internal energy.



(a) relative mass conservation losses



(b) relative momentum conservation losses



(c) relative energy conservation losses

FIG. 4.6. Relative conservation losses for test case 3.

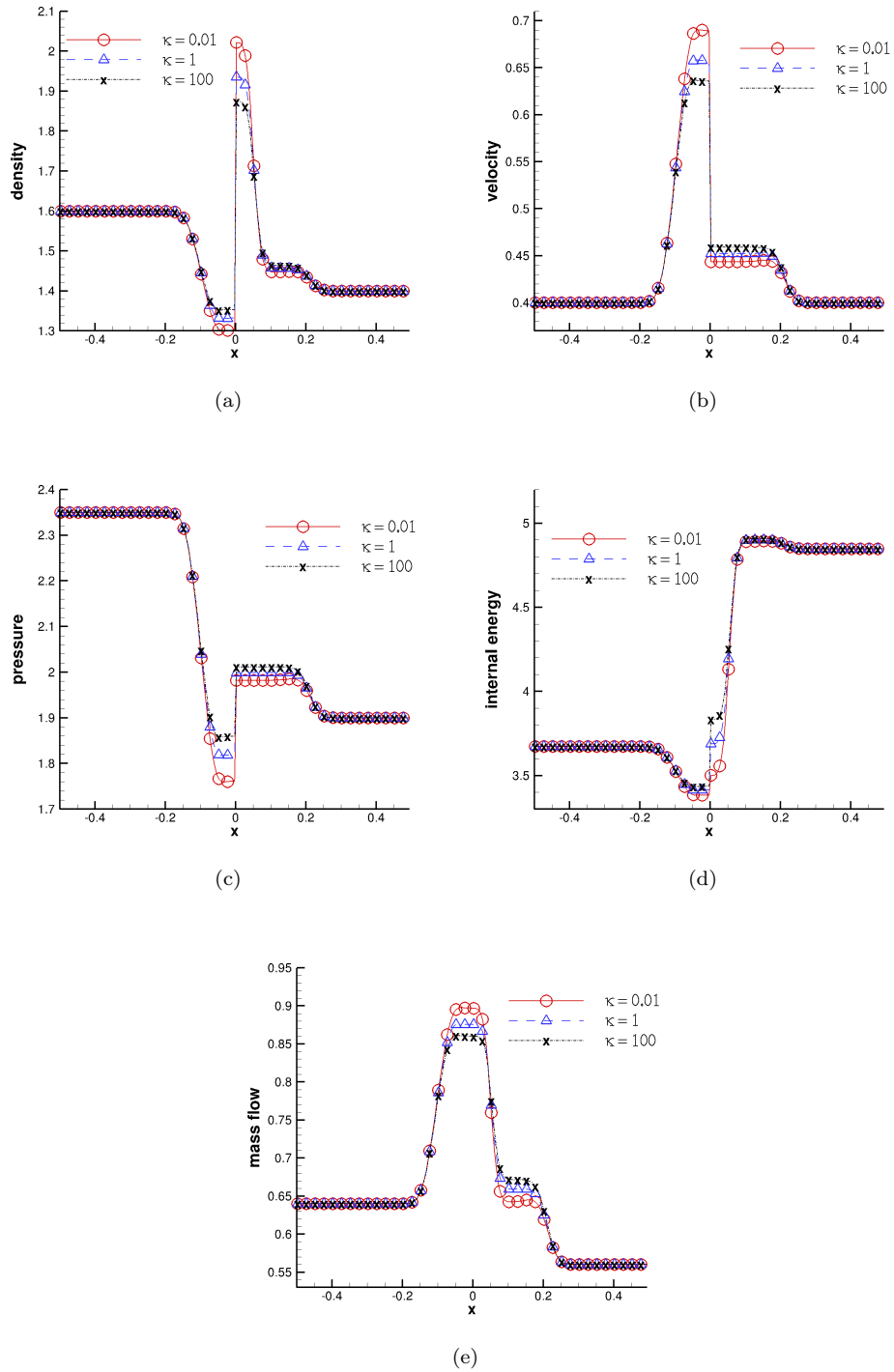
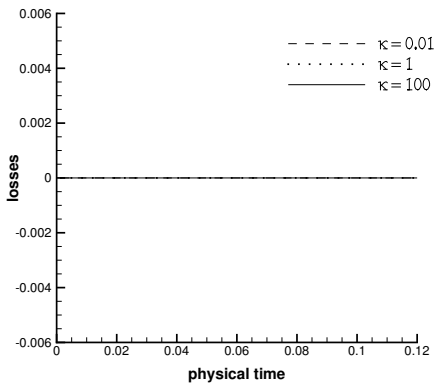
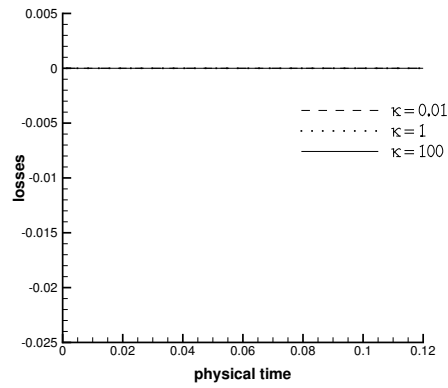


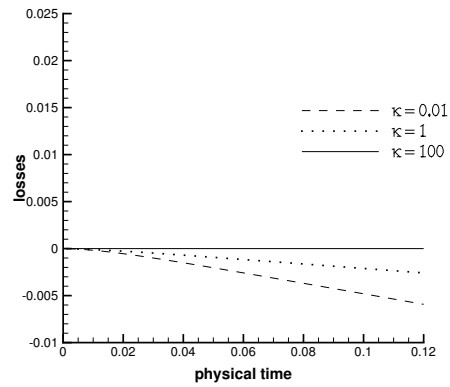
FIG. 4.7. Constrained optimization for test case 4: profiles of primitive variables (ρ, u, p), mass flow and internal energy.



(a) relative mass conservation losses



(b) relative momentum conservation losses



(c) relative energy conservation losses

FIG. 4.8. Relative conservation losses for test case 4.

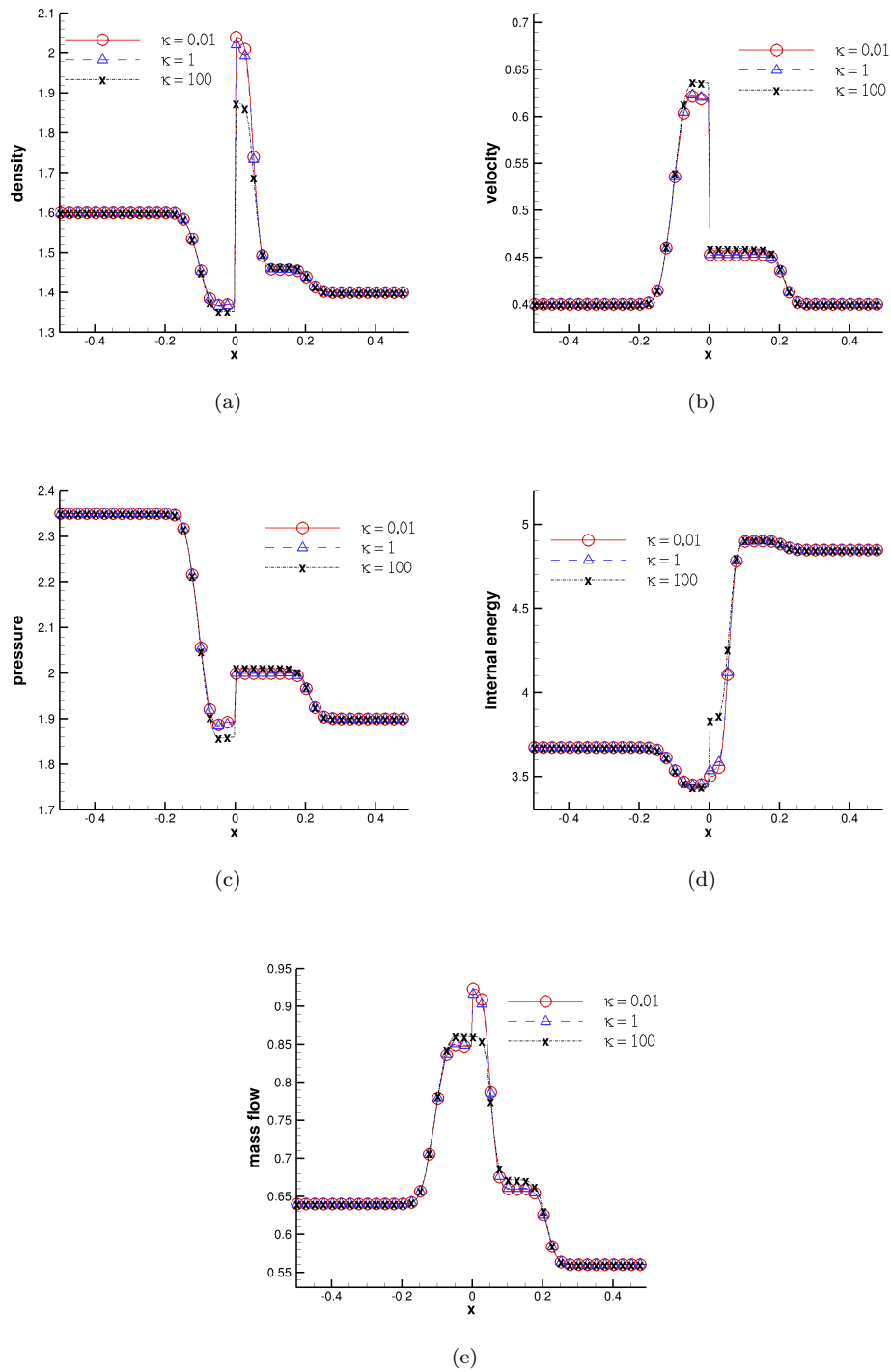
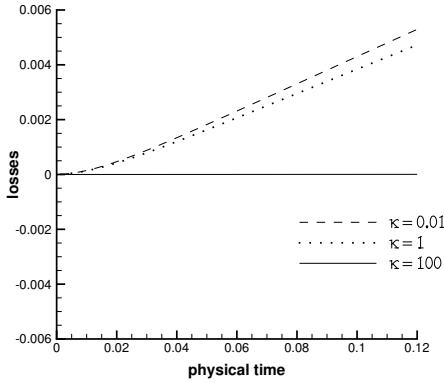
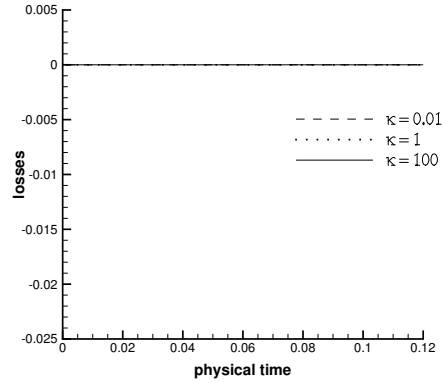


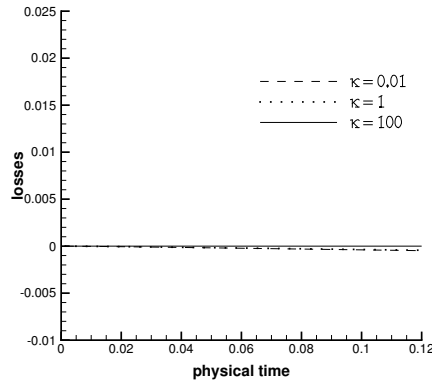
FIG. 4.9. Constrained optimization for test case 5: profiles of primitive variables (ρ, u, p), mass flow and internal energy.



(a) relative mass conservation losses



(b) relative momentum conservation losses



(c) relative energy conservation losses

FIG. 4.10. *Relative conservation losses for test case 5.*

5. Concluding remarks

We have extended the study of interface coupling with local measure source term [4, 34, 5] to the general case where all the source terms are non zero. An approximate Riemann solver preserving equilibrium solutions has been proposed for the numerical approximation of the coupled model in the context of the gas dynamics equations with different closure laws for the pressure, as a model example. The construction of this solver relies on a relaxation approximation where the associated Riemann problem has been solved analytically. This relaxation method has been used for the evaluation of optimal source terms satisfying physical criteria at the coupling interface. In the context of constrained optimization problems, numerical experiments have shown that it is possible to impose physical quantities such as mass flow or temperature at the coupling interface while controlling conservation losses. Future investigations should include an extension to several space dimensions and adaptive procedures, together with the coupling of more realistic models (even if we cannot expect explicit theoretical results for general models, at least it is interesting to extend the numerical approach

to less academic cases).

Appendix A. Mathematical properties of the cost functions \mathcal{J}_1 and \mathcal{J}_2 .

First, recall that the sequence of the gradient projection algorithm converges to the optimum if the objective function \mathcal{J} is $\alpha_{\mathcal{J}}$ -convex differentiable and $\nabla\mathcal{J}$ is $\mathcal{C}_{\nabla\mathcal{J}}$ -Lipschitz continuous. More precisely, if the algorithm parameter $\mu_{\mathcal{J}}$ is chosen such that the following inequality holds

$$0 < \mu_{\mathcal{J}} < \frac{2\alpha_{\mathcal{J}}}{\mathcal{C}_{\nabla\mathcal{J}}^2},$$

then the algorithm converges.

Here, we give some properties of the cost functions \mathcal{J}_1 and \mathcal{J}_2 respectively defined in (4.2) and (4.3).

PROPOSITION A.1. *For all control parameter $\kappa > 0$, the functions \mathcal{J}_1 and \mathcal{J}_2 are strictly convex. For each $\kappa > 0$, set:*

$$\alpha_1 = \frac{1}{2} \left[\text{Tr}(\text{Hess}(\mathcal{J}_1)) - \sqrt{\text{Tr}(\text{Hess}(\mathcal{J}_1))^2 - 4\det(\text{Hess}(\mathcal{J}_1))} \right] > 0,$$

$$\mathcal{C}_{\nabla\mathcal{J}_1} = \text{rspec}(\text{Hess}(\mathcal{J}_1)) > 0,$$

$$\alpha_2 = \frac{2\kappa^2}{a^6\tau_R^4} > 0, \quad \mathcal{C}_{\nabla\mathcal{J}_2} = \frac{2}{(\varepsilon_{ref}m_+)^2} + \alpha_2 > 0,$$

where for a matrix A , $\text{Tr}(A)$ denotes its trace, $\det(A)$ its determinant and $\text{rspec}(A)$ its spectral radius. $\text{Hess}(\mathcal{J}_1)$ denotes the Hessian matrix of the cost function \mathcal{J}_1 . Then, for $i=1,2$, there exists $\alpha_{\mathcal{J}_i}$ strictly positive and strictly smaller than α_i such that the function \mathcal{J}_i is $\alpha_{\mathcal{J}_i}$ -convex differentiable and $\nabla\mathcal{J}_i$ is $\mathcal{C}_{\nabla\mathcal{J}_i}$ -Lipschitz. Finally, for $i=1,2$, the gradient projection algorithm converges for all parameter $\mu_{\mathcal{J}_i}$ verifying:

$$0 < \mu_{\mathcal{J}_i} < \frac{2\alpha_{\mathcal{J}_i}}{\mathcal{C}_{\nabla\mathcal{J}_i}^2}.$$

Proof. First of all, one can check that $\text{Tr}(\text{Hess}(\mathcal{J}_1)) > 0$ and $\det(\text{Hess}(\mathcal{J}_1)) > 0$ for all $\kappa > 0$, since after easy calculations we get:

$$\text{Tr}(\text{Hess}(\mathcal{J}_1)) = \frac{2(a\tau_L^* - u^*)^2}{[am_{ref}(\tau_L^* + \tau_R^*)]^2} + \frac{2\kappa^2}{a^2} + \frac{2}{[am_{ref}(\tau_L^* + \tau_R^*)]^2} + \frac{2\kappa^2}{a^4\tau_R^2} > 0,$$

and

$$\det(\text{Hess}(\mathcal{J}_1)) = \frac{4\kappa^2}{a^4} \left[\frac{(a\tau_L^* - u^*)^2}{[am_{ref}\tau_R(\tau_L^* + \tau_R^*)]^2} + \frac{1}{[m_{ref}(\tau_L^* + \tau_R^*)]^2} + \frac{\kappa^2}{a^2\tau_R^2} \right] > 0.$$

This implies in particular the strict convexity of \mathcal{J}_1 since its Hessian matrix is positive definite. Let

$$\Delta = \text{Tr}(\text{Hess}(\mathcal{J}_1))^2 - 4\det(\text{Hess}(\mathcal{J}_1)). \quad (1.1)$$

First, remark that:

$$\begin{aligned}\Delta &= \left[\partial_{\mathcal{M}_\rho}^2 \mathcal{J}_1 + \partial_{\mathcal{M}_{\rho u}}^2 \mathcal{J}_1 \right]^2 - 4 \left[\partial_{\mathcal{M}_\rho}^2 \mathcal{J}_1 \partial_{\mathcal{M}_{\rho u}}^2 \mathcal{J}_1 - \left(\partial_{\mathcal{M}_\rho, \mathcal{M}_{\rho u}}^2 \mathcal{J}_1 \right)^2 \right] \\ &= \left[\partial_{\mathcal{M}_\rho}^2 \mathcal{J}_1 - \partial_{\mathcal{M}_{\rho u}}^2 \mathcal{J}_1 \right]^2 + 4 \left(\partial_{\mathcal{M}_\rho, \mathcal{M}_{\rho u}}^2 \mathcal{J}_1 \right)^2 > 0,\end{aligned}$$

since

$$\partial_{\mathcal{M}_\rho, \mathcal{M}_{\rho u}}^2 \mathcal{J}_1 = \frac{2(a\tau_L^* - u^*)}{[am_{ref}(\tau_L^* + \tau_R^*)]^2} \neq 0 \quad (\text{because } \lambda_1^r < 0),$$

and then observe that:

$$\Delta - \text{Tr}(\text{Hess}(\mathcal{J}_1))^2 = -4 \det(\text{Hess}(\mathcal{J}_1)) < 0.$$

From $\Delta > 0$, $\text{Tr}(\text{Hess}(\mathcal{J}_1)) > 0$ and the previous inequality, we deduce that the term α_1 defined in the present proposition is strictly positive. So, there exists $\alpha_{\mathcal{J}_1}$ such that $0 < \alpha_{\mathcal{J}_1} < \alpha_1$. Now, in order to show that \mathcal{J}_1 is $\alpha_{\mathcal{J}_1}$ -convex, let us define the function

$$f_1: (\mathcal{M}_\rho, \mathcal{M}_{\rho u}) \in \mathbb{R}^2 \mapsto \mathcal{J}_1(\mathcal{M}_\rho, \mathcal{M}_{\rho u}) - \frac{\alpha_{\mathcal{J}_1}}{2} \|(\mathcal{M}_\rho, \mathcal{M}_{\rho u})\|^2 \in \mathbb{R},$$

where $\|\cdot\|$ denotes the usual Euclidean norm on \mathbb{R}^2 . After easy calculations, we obtain

$$\text{Tr}(\text{Hess}(f_1)) = \text{Tr}(\text{Hess}(\mathcal{J}_1)) - 2\alpha_{\mathcal{J}_1},$$

and

$$\det(\text{Hess}(f_1)) = \det(\text{Hess}(\mathcal{J}_1)) - \alpha_{\mathcal{J}_1} \text{Tr}(\text{Hess}(\mathcal{J}_1)) + \alpha_{\mathcal{J}_1}^2.$$

Since $\alpha_{\mathcal{J}_1} < \alpha_1 < \text{Tr}(\text{Hess}(\mathcal{J}_1))/2$, it is clear that the trace of the Hessian matrix of the function f_1 is strictly positive. Moreover, considering the following polynomial of degree 2 in X :

$$\mathcal{P}(X) = X^2 - \text{Tr}(\text{Hess}(\mathcal{J}_1))X + \det(\text{Hess}(\mathcal{J}_1)),$$

one can see that its discriminant is given by $\Delta > 0$ in (1.1) and α_1 is its smallest root. Since $\alpha_{\mathcal{J}_1} < \alpha_1$, the determinant of the Hessian matrix of the function f_1 is strictly positive and we deduce that f_1 is in particular convex that is to say that \mathcal{J}_1 is $\alpha_{\mathcal{J}_1}$ -convex. For all $\kappa > 0$, the gradient $\nabla \mathcal{J}_1$ of the function \mathcal{J}_1 is Lipschitz-continuous with constant $\mathcal{C}_{\nabla \mathcal{J}_1}$ since the eigenvalues of the Hessian of \mathcal{J}_1 are bounded above by $\mathcal{C}_{\nabla \mathcal{J}_1}$. The constant $\mathcal{C}_{\nabla \mathcal{J}_1}$ is strictly positive because \mathcal{J}_1 is strictly convex and is independent of the variables \mathcal{M}_ρ and $\mathcal{M}_{\rho u}$ because the Hessian is independent of these variables.

Now, let us introduce $0 < \alpha_{\mathcal{J}_2} < \alpha_2$ and the function

$$f_2: \mathcal{M}_{\rho E} \in \mathbb{R} \mapsto \mathcal{J}_2(\mathcal{M}_{\rho E}) - \frac{\alpha_{\mathcal{J}_2}}{2} \mathcal{M}_{\rho E}^2 \in \mathbb{R}.$$

Then, the second order derivative of \mathcal{J}_2 is given by

$$\mathcal{J}_2''(\mathcal{M}_{\rho E}) = \begin{cases} \frac{2\kappa^2}{a^6 \tau_R^4} & \text{if } \mathcal{M}_{\rho E} \in \mathcal{D}_{adm}^-, \\ \frac{2\kappa^2}{a^6 \tau_R^4} + \frac{2}{\varepsilon_{ref}^2 m_+^2} & \text{if } \mathcal{M}_{\rho E} \in \mathcal{D}_{adm}^+, \end{cases}$$

which implies $f_2''(\mathcal{M}_{\rho E}) = \mathcal{J}_2''(\mathcal{M}_{\rho E}) - \alpha_{\mathcal{J}_2} > 0$, that is to say that the function \mathcal{J}_2 is $\alpha_{\mathcal{J}_2}$ -convex. Moreover, given the above relations, we directly obtain

$$\sup |\mathcal{J}_2''(\mathcal{M}_{\rho E})| \leq \mathcal{C}_{\nabla \mathcal{J}_2},$$

and applying a corollary of the mean value theorem, we deduce that for all $\kappa > 0$, the gradient $\nabla \mathcal{J}_2$ is Lipschitz-continuous with constant $\mathcal{C}_{\nabla \mathcal{J}_2}$. This concludes the proof. \square

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